

# **Practice Course in Plasma**

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**Institute of Space and Plasma Sciences, National Cheng Kung University**

**2021 spring semester**

**Thursday 9:10-12:00**

**Material: <http://capst.ncku.edu.tw/PGS/index.php/teaching/>**

**Lecture 6**

# Reference

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- 真空技術與應用, 國家實驗研究院儀器科技研究中心出版

# Vacuum Levels

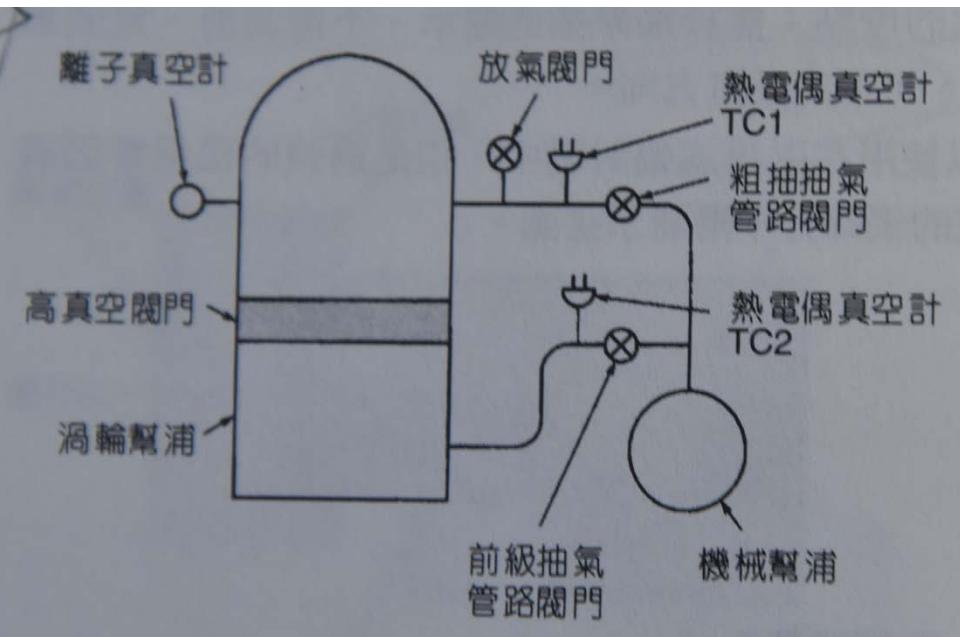


Levels	Pressure	Mean Free Path ( $\lambda$ ) vs System size (d)	Flow type
Rough Vacuum (粗略真空)	$10^{-1} \sim 760$ Torr	$\lambda \ll d$	Viscos flow
Medium Vacuum (中度真空)	$10^{-5} \sim 10^{-1}$ Torr	$\lambda \sim d$	Transition flow
High Vacuum (高真空)	$10^{-9} \sim 10^{-5}$ Torr	$\lambda \gg d$	Molecular flow
Ultra High Vacuum (超高真空)	$< 10^{-9}$ Torr	$\lambda \gg d$	Molecular flow

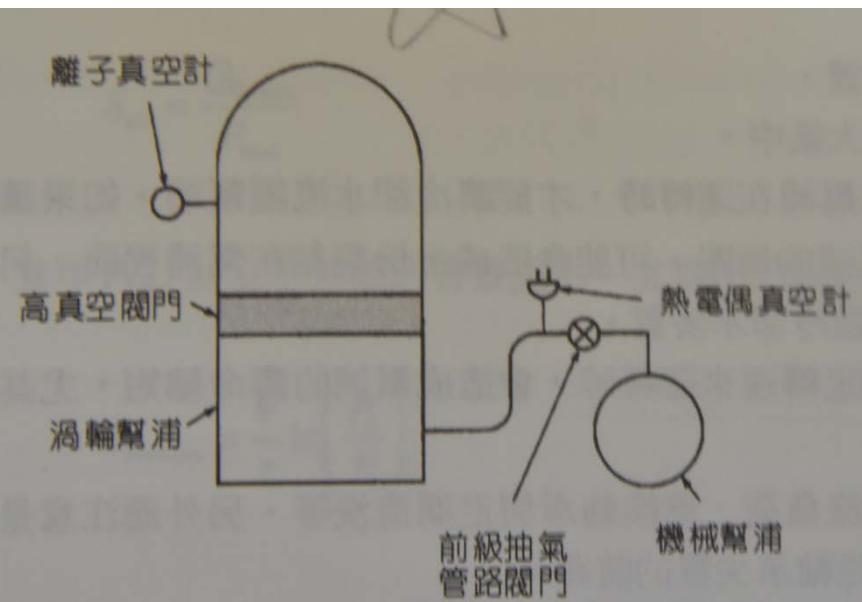
# 渦輪真空幫浦系統



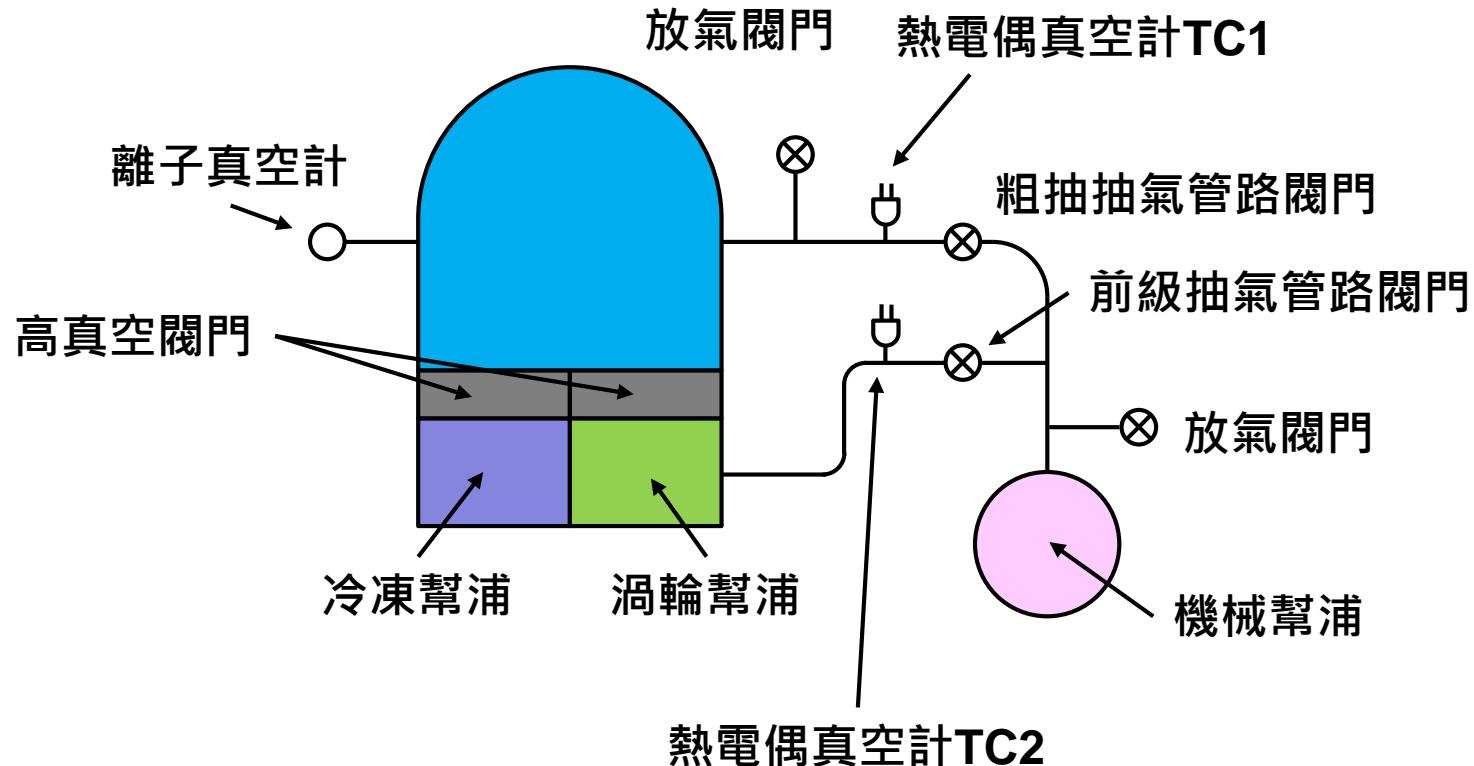
- 有預抽管



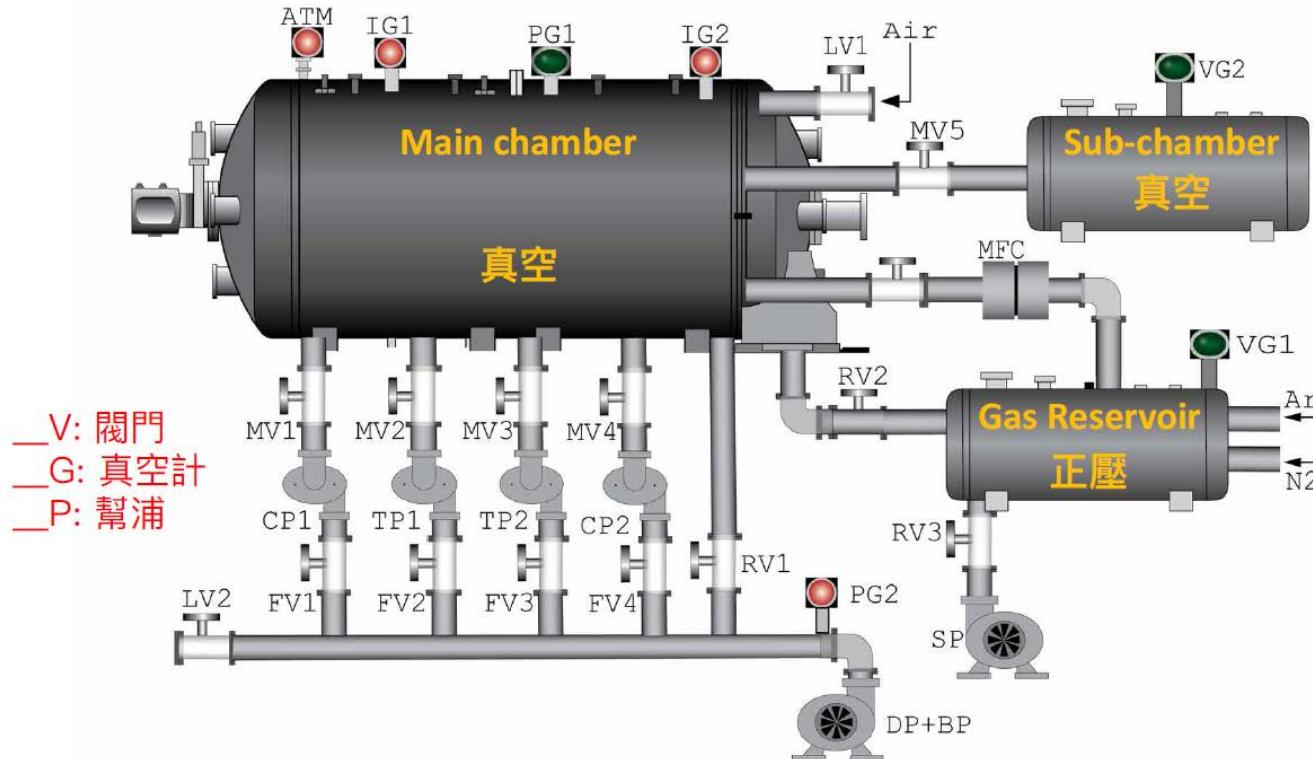
- 無預抽管



# 冷凍幫浦系統



# SPOC



(5) ( ) 30% 抽真空開關的順序：

- (a) 開FV2; (b) 關RV1; (c) 開TP1; (d) 開DP+BP; (e) 開RV1; (f) 開MV2.

(6) ( ) 30% 破真空開關的順序：

- (a) 關FV2; (b) 關DP+BP; (c) 聽到“搭”一聲; (d) 開LV1; (e) 關LV2;
- (f) 關MV2; (g) 開艙; (h) 關RV1; (i) 關TP1; (j) 開LV2.

# Space Plasma Operation Chamber (SPOC)



- Training of using SPOC is scheduled for next lecture (3/25)

# 各種真空系統相關功能之比較



	離子幫浦系統	擴散幫浦系統	渦輪分子幫浦系統	冷凍幫浦系統
抽氣速率	低	高	中	中
終極壓力	$< 10^{-9}$ Torr	$10^{-6} \sim 10^{-9}$ Torr	$10^{-9}$ Torr	$10^{-9}$ Torr
使用難易	簡單	普通	簡單	普通
使用壽命	有限時間	長久	長久	再生後使用
系統潔淨度	好	差	好	好
系統震動	最小	小	差	小
維護難易	容易	容易	難	難
裝置成本	高	低	高	高
運轉成本	低	高	中	中

# 真空系統結構元件要求



- 機械強度能承受壓差所產生的外力。
- 氣密佳，不應是多孔結構，不能有裂縫、小孔等成為洩漏通道的缺陷。
- 低逸氣率與低滲透率。
- 在工作溫度及烘烤溫度下的蒸氣要低。
- 化學隱定性佳，不易氧化、耐腐蝕。
- 在一定溫度範圍內保持其真空性能及機械強度。
- 加工容易、銲接性佳。
- Ex : 不鏽鋼、無氧高導銅、碳鋼、鋁
- Not good : 鋅、鉛、鎘、銓、鋰、鉀、硒、硫、...
- 氢氣滲透率 : 鋁 < 鉛 < 銅 < 鈀 < 鐵 < 鎳 < 鈀

# 常見橡膠與塑膠真空材料逸氣率



材料種類		$q_1$ ( $10^{-5}$ W/cm $^2$ )	$\alpha_1$	$q_{10}$ ( $10^{-5}$ W/cm $^2$ )	$\alpha_{10}$
橡 膠	Butyl DR 41	200	0.68	53	0.64
	Neoprene	4000	0.4	2400	0.4
	Perbunan	467	0.3	293	0.5
	Silicone	930	—	267	—
	Viton A (fresh)	152	0.8	—	—
	Viton A (bake 12h at 200 °C)	—	—	0.27	—
塑 膠	Polyimide (bake 12h at 300 °C)	—	—	0.005	—
	Araldite (molded)	155	0.8	47	0.8
	Araldite D	253	0.3	167	0.5
	Araldite F	200	0.5	97	0.5
	Kel-F	5	0.57	2.3	0.53
	Methyl Methacrylate	560	0.9	187	0.57
	Mylar (24h at 95% RH)	307	0.75	53	—
	Nylon	1600	0.5	800	0.5
	Plexiglas	961	0.44	36	0.44
	Plexiglas	413	0.4	240	0.4
Polyester-glass Laminate		333	0.84	107	0.81
Polystyrene		2667	1.6	267	1.6
PTFE		40	0.45	26	0.56
PVC (24h at 95% RH)		113	1.0	2.4	—
Teflon		8.7	0.5	3.3	0.2

# 常見高分子真空材料之氣體滲透率



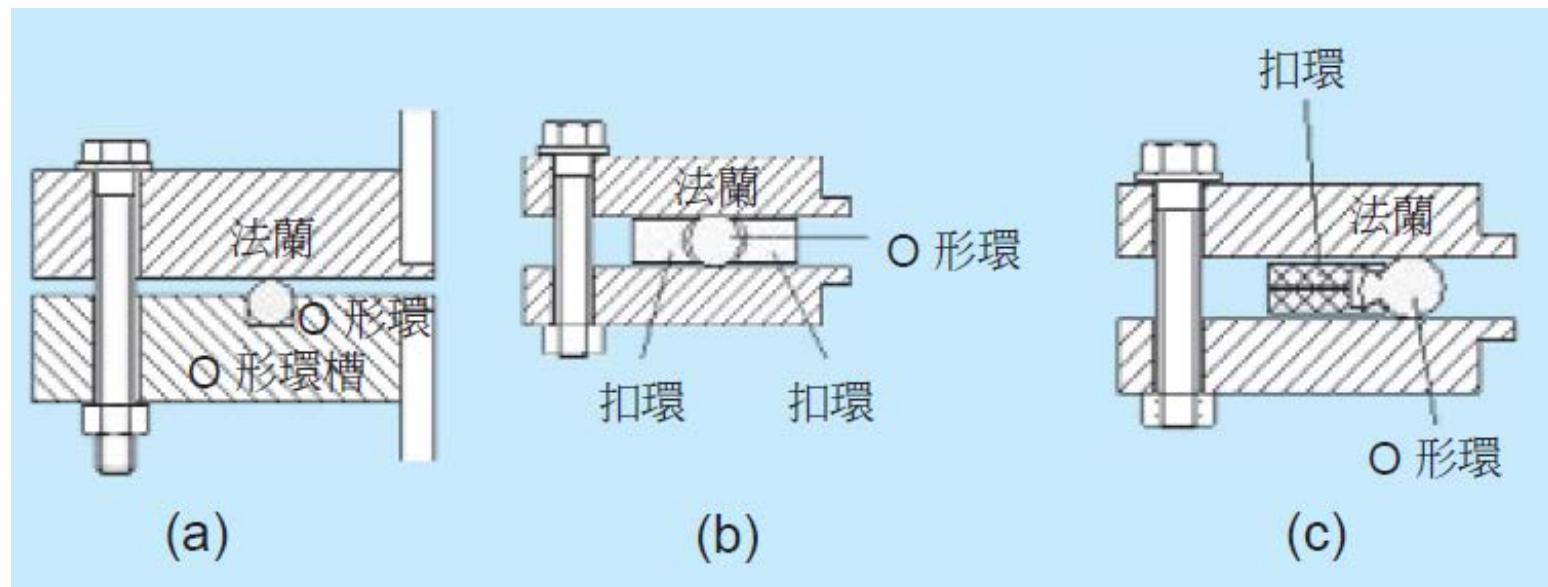
材料種類	滲透率 ( $10^{-12} \text{ m}^2/\text{s}$ )					
	N <sub>2</sub>	O <sub>2</sub>	H <sub>2</sub>	He	H <sub>2</sub> O	CO <sub>2</sub>
PTFE	2.5	8.2	20	570	—	—
Perspex	—	—	2.7	5.7	—	—
Nylon 31	—	—	0.13	0.3	—	—
Neoprene CS2368B	0.21	1.5	8.2	7.9	—	—
Viton A	0.05	1.1	2.2	8.9	—	5.9
Kapton	0.03	0.1	1.1	1.9	—	0.2
Buna-S	4.8	—	—	—	—	940.0
Perbunan	0.8	—	—	—	—	23.0
Delrin	—	48.0	—	—	17.0	93.0
Kel-F	0.99	0.46	—	—	0.22	—

# 橡膠材料極限真空試驗結果



材料	極限真空度 (Torr)	極限真空度 (Torr)
	法蘭溫度 6°C 時	法蘭溫度 25°C 時
天然橡膠	$4.5 \times 10^{-9}$	$1.2 \times 10^{-9}$
丁基橡膠	$1.0 \times 10^{-9}$	$1.75 \times 10^{-10}$
氯丁橡膠	$2.1 \times 10^{-9}$	$2.1 \times 10^{-10}$
硝酸基橡膠	$3.8 \times 10^{-9}$	$4.8 \times 10^{-10}$
矽膠 (紅)	$2.2 \times 10^{-9}$	—
矽膠 (紅)	$3.2 \times 10^{-9}$	—
維通 (綠)	$1.3 \times 10^{-9}$	$5.6 \times 10^{-10}$
聚四氟乙烯	$4.2 \times 10^{-9}$	$1.0 \times 10^{-9}$

# 可拆卸真空封合 – O-ring



# O-ring groove design



特 徵 尺 寸					
A	$1.15d$	$1.4d$	$1.4d$	$0.9d - 0.95d$	$d$
B	$0.72d$	$0.7d$	$0.7d$	$0.75d - 0.8d$	$1.15d - 1.3d$
R	$0.15d - 0.22d$ 圓角磨光 $R_a < 1.6 \mu m$				

# More clamp

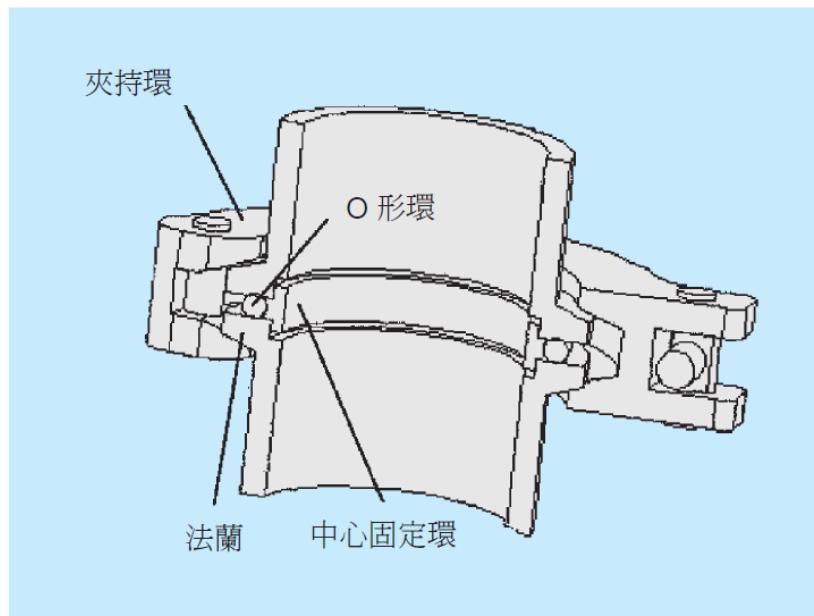


圖 3. KF 法蘭封合結構示意圖。

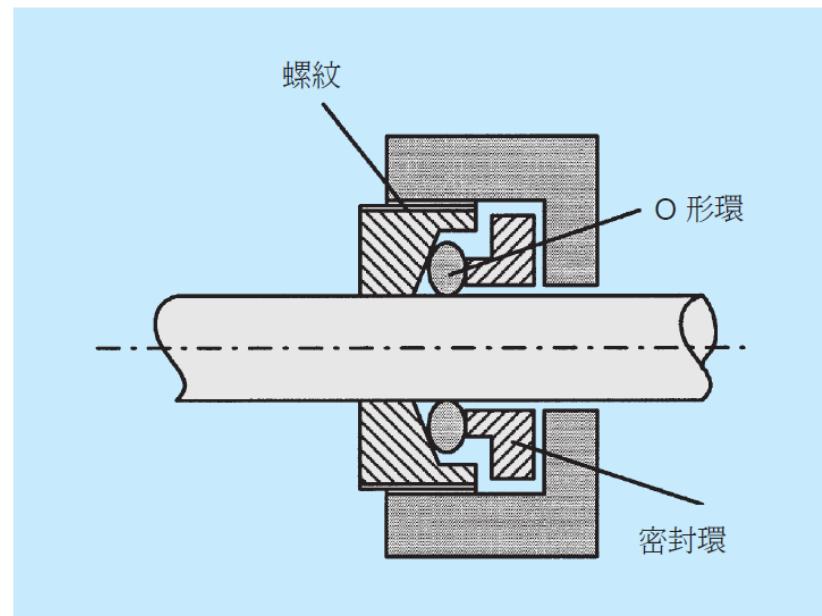
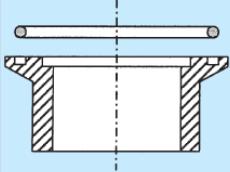
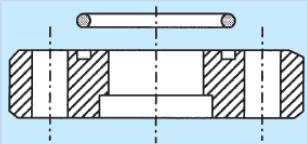


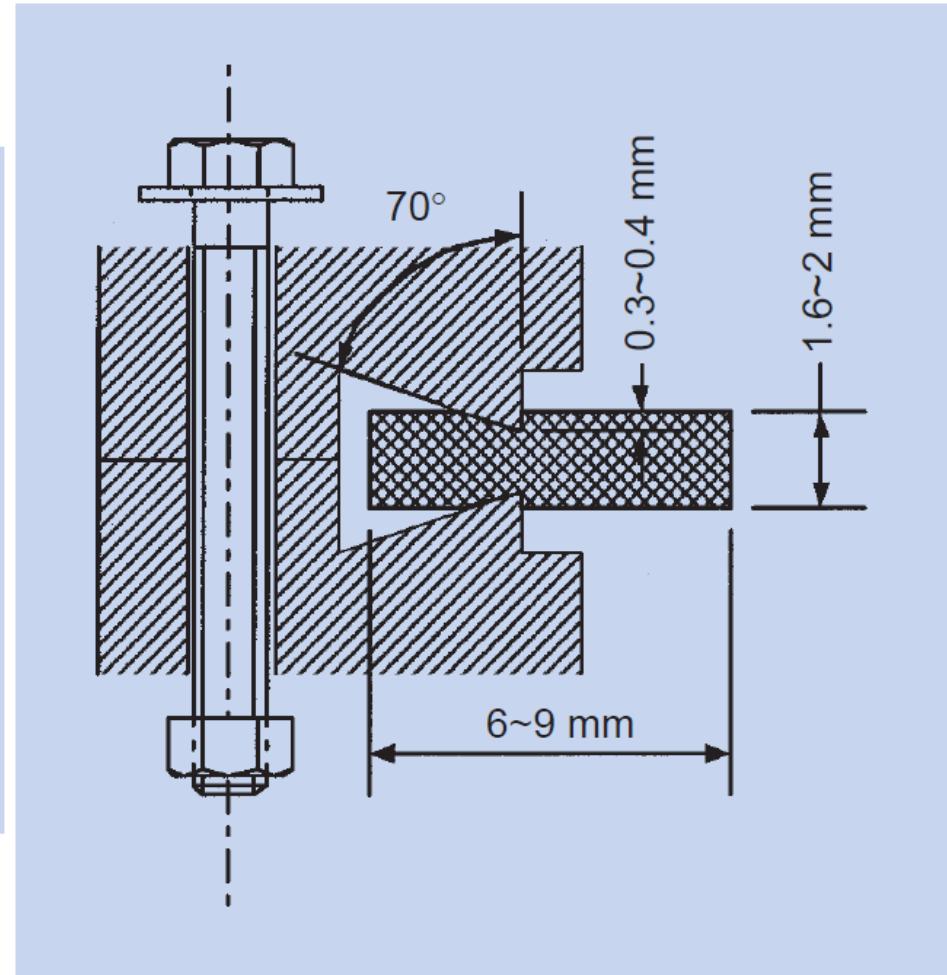
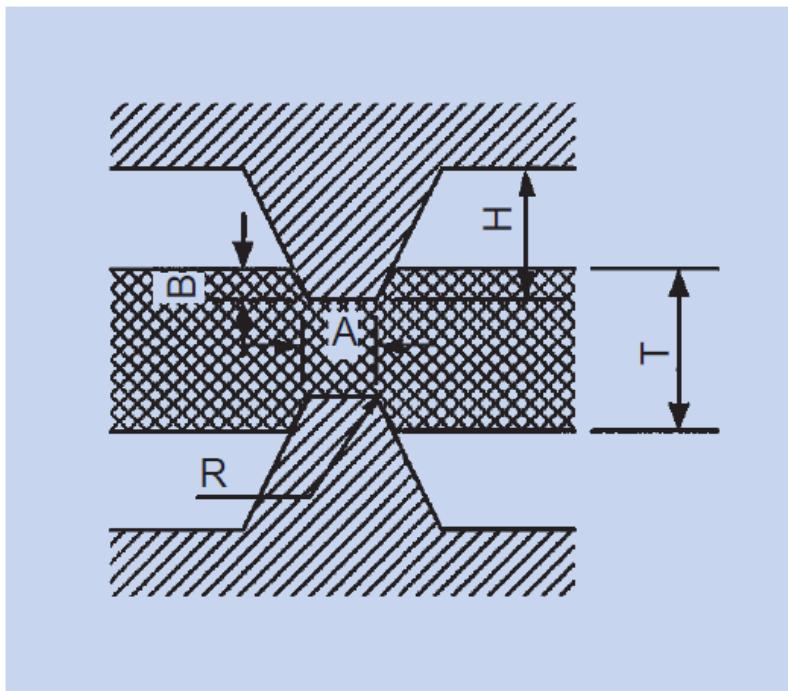
圖 4. 管路之錐形壓縮封合。

# Comparison of different types of flange

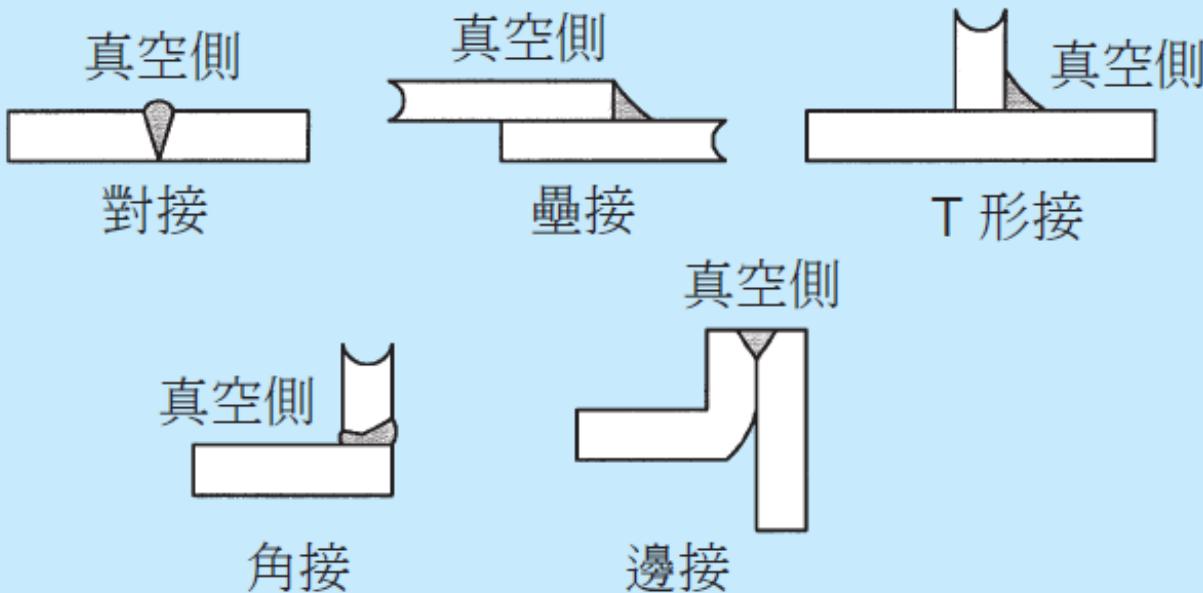


法蘭種類	適用系統	特性
 ISO 法蘭	中低真空系統及無需烘烤之高真空系統 (壓力大於 $10^{-8}$ Torr 之系統使用) 使用 Viton O 形環烘烤至 200 °C，操作溫度可達 150 °C。	高分子封合材料 O 形環，可重覆使用，可以使用固定中心環和平面法蘭或在法蘭上以環槽固定 O 形環，組裝拆卸快速、成本較為經濟。 組裝時可使用真空油脂輕輕塗覆 O 形環，可以提高封合性能。 小管徑使用夾緊環 (hing clamp)，手動鎖緊即可，較為方便，大管徑則使用緊固扣環 (claw clamp)。
 CF 法蘭 (conflat 法蘭)	超高真空系統封合 (壓力小於 $10^{-8}$ Torr 之系統使用)。可使用金屬墊圈及 Viton O 形環，若使用 Viton O 形環可烘烤至 200 °C，操作溫度可達 150 °C。	金屬墊圈封合材料以及 Viton O 形環，封合滲漏很微小。 法蘭刀口及封合面的尺寸精度及表面粗糙度要求高。 需依要求進行清潔與螺絲組裝程序。
 ASA-ANSI 法蘭	中低真空系統及無需烘烤之高真空系統 (壓力大於 $10^{-8}$ Torr 之系統使用) 使用 Viton O 形環可烘烤至 200 °C，操作溫度可達 150 °C。	高分子封合材料 O 形環，封合效果較 ISO 法蘭佳。 可熔接或硬焊於腔體或元件需依要求進行清潔與螺絲組裝程序。

# 金屬墊圈刀刃及斜楔法蘭封合



# 永久封合

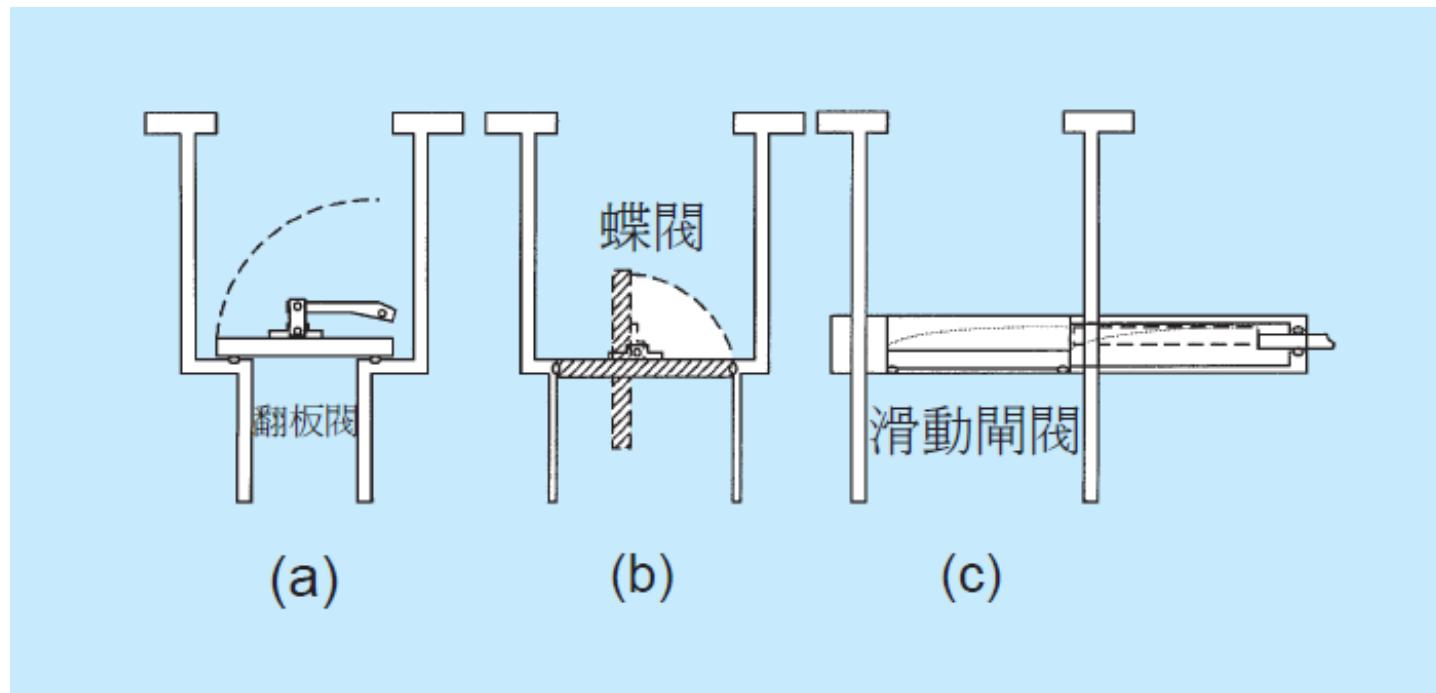


# 真空閥門分類

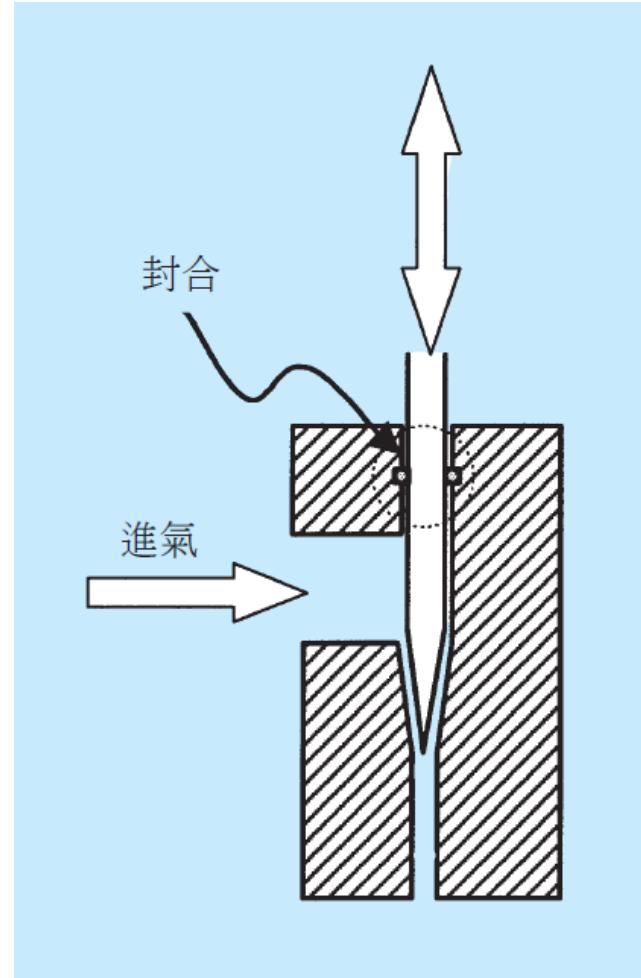
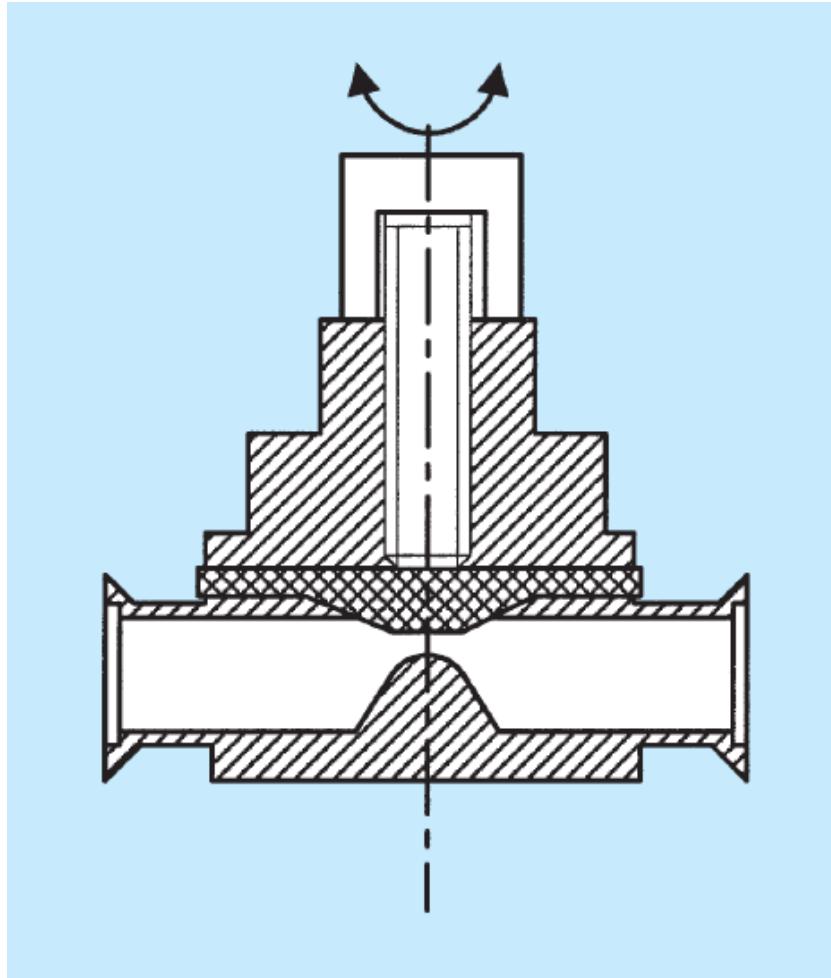


分類依據	閥門名稱
工作壓力	中低真空閥、高真空閥、超高真空閥
用途	截止閥、隔絕閥、放氣閥、節流閥、換向閥、封閉送料閥
驅動方式	手動閥、電動閥、手電兩用閥、電磁閥、氣動閥、液壓式真空閥
材料	玻璃龍頭閥、金屬真空閥
結構特點	擋板閥、翻板閥、蝶閥、連桿閥、隔板閥、閘閥、雙通閥、三通閥、四通閥、直通閥、角閥

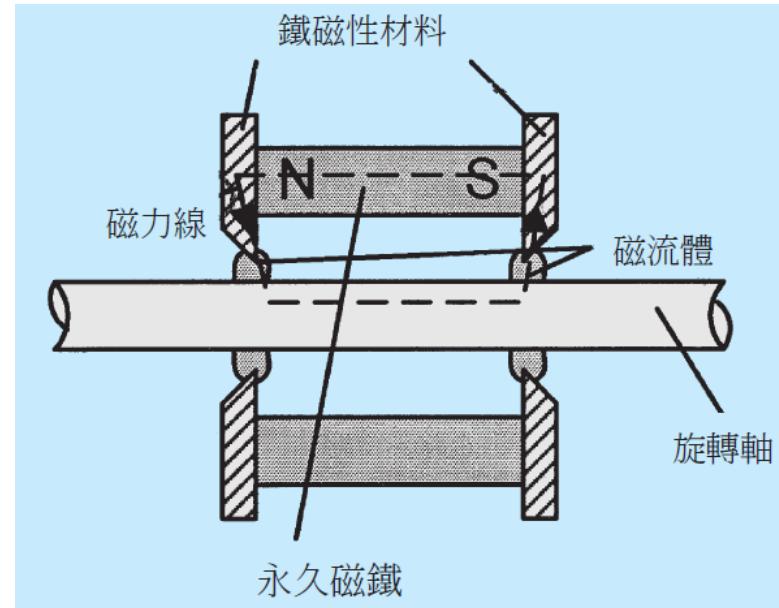
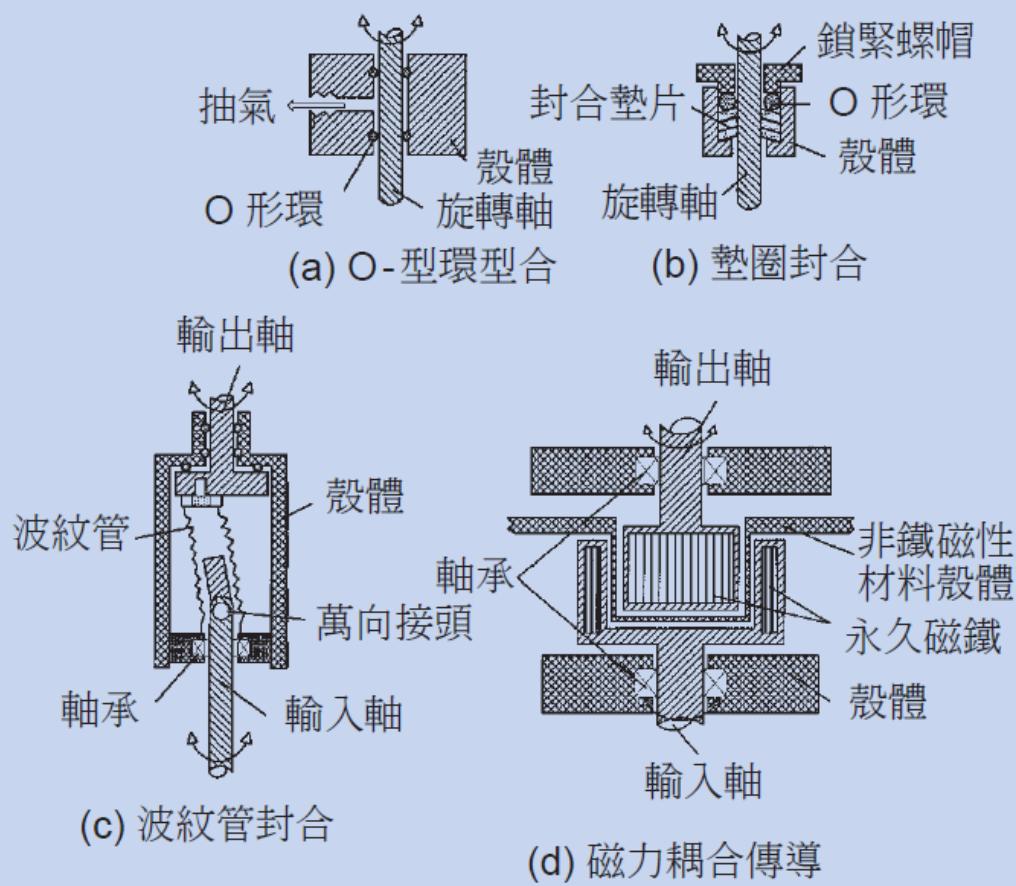
# 閘閥結構



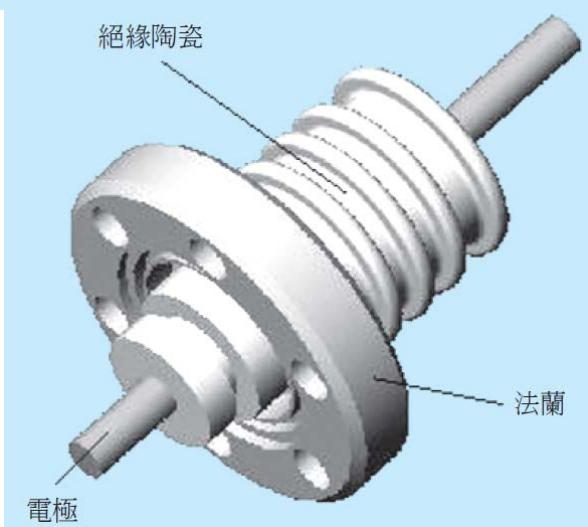
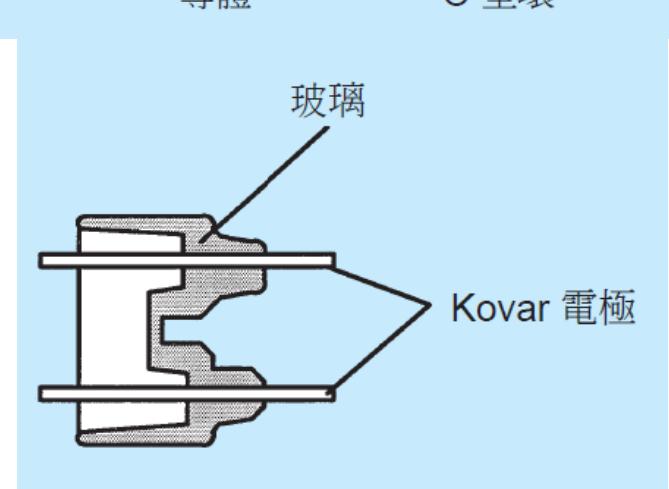
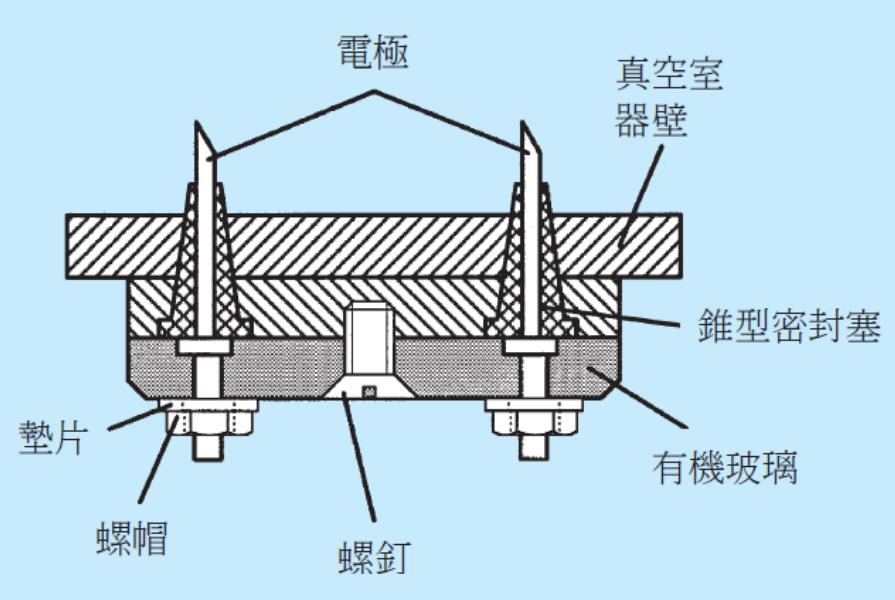
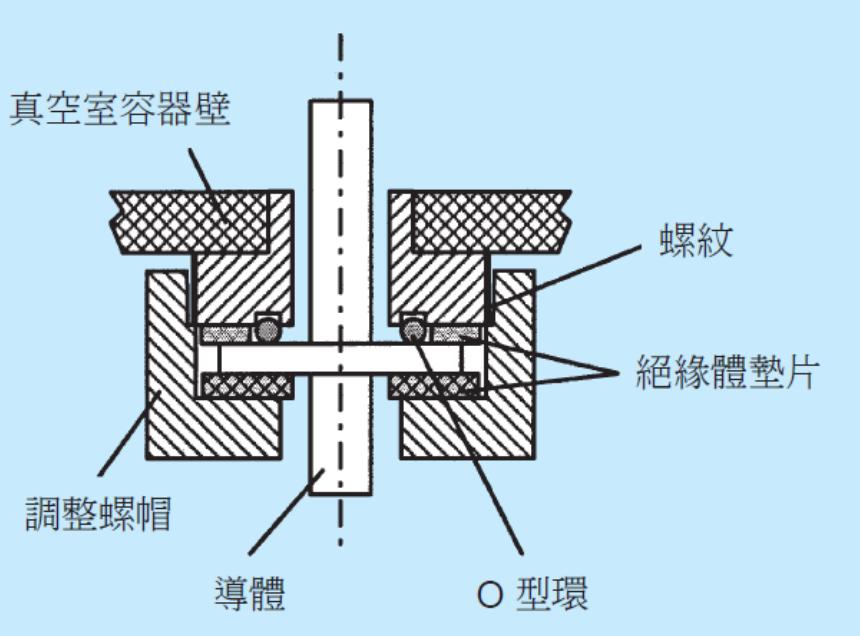
# 可調流量閥門 - 薄膜真空閥、針閥



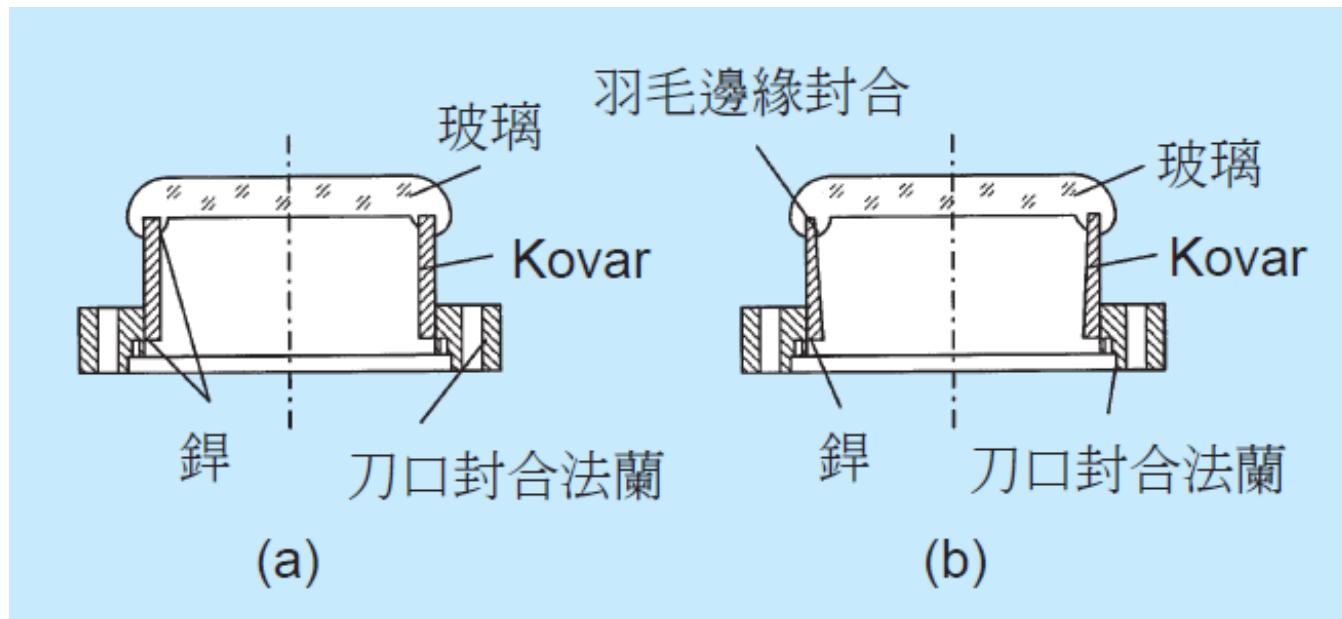
# 機械運動引入(feedthrough)



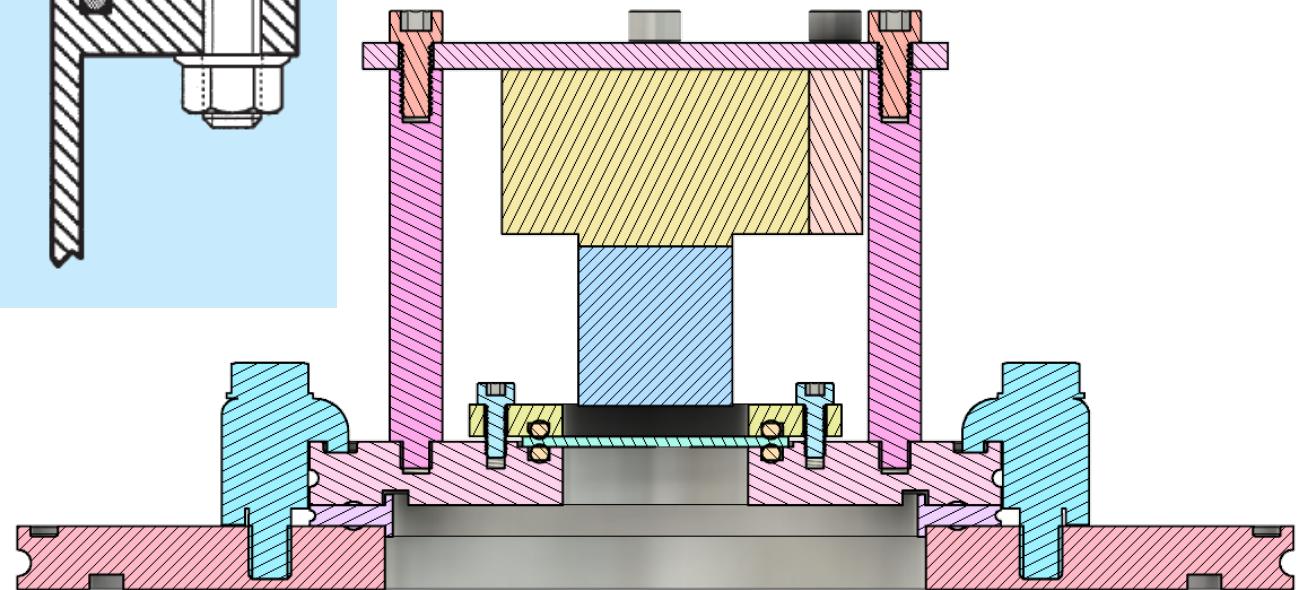
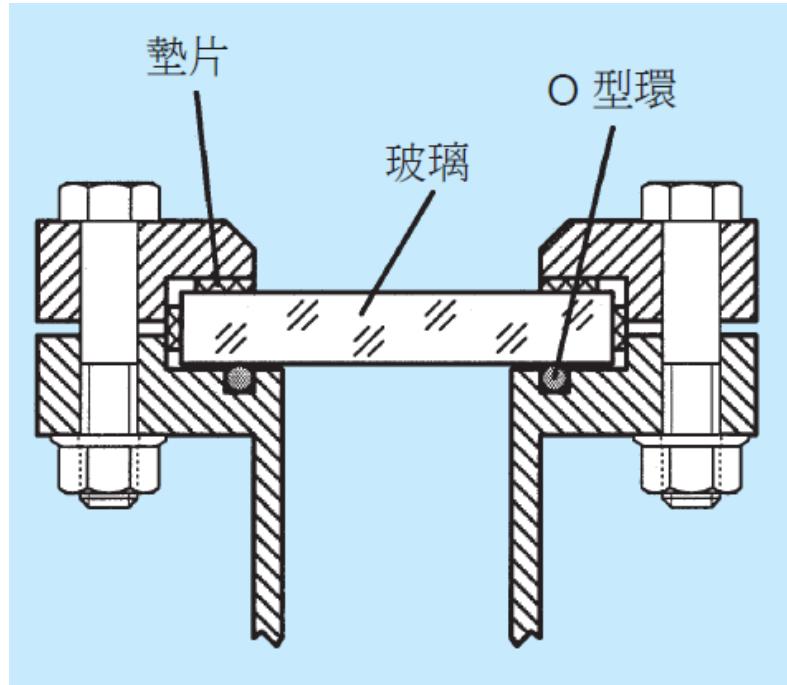
# 電引入



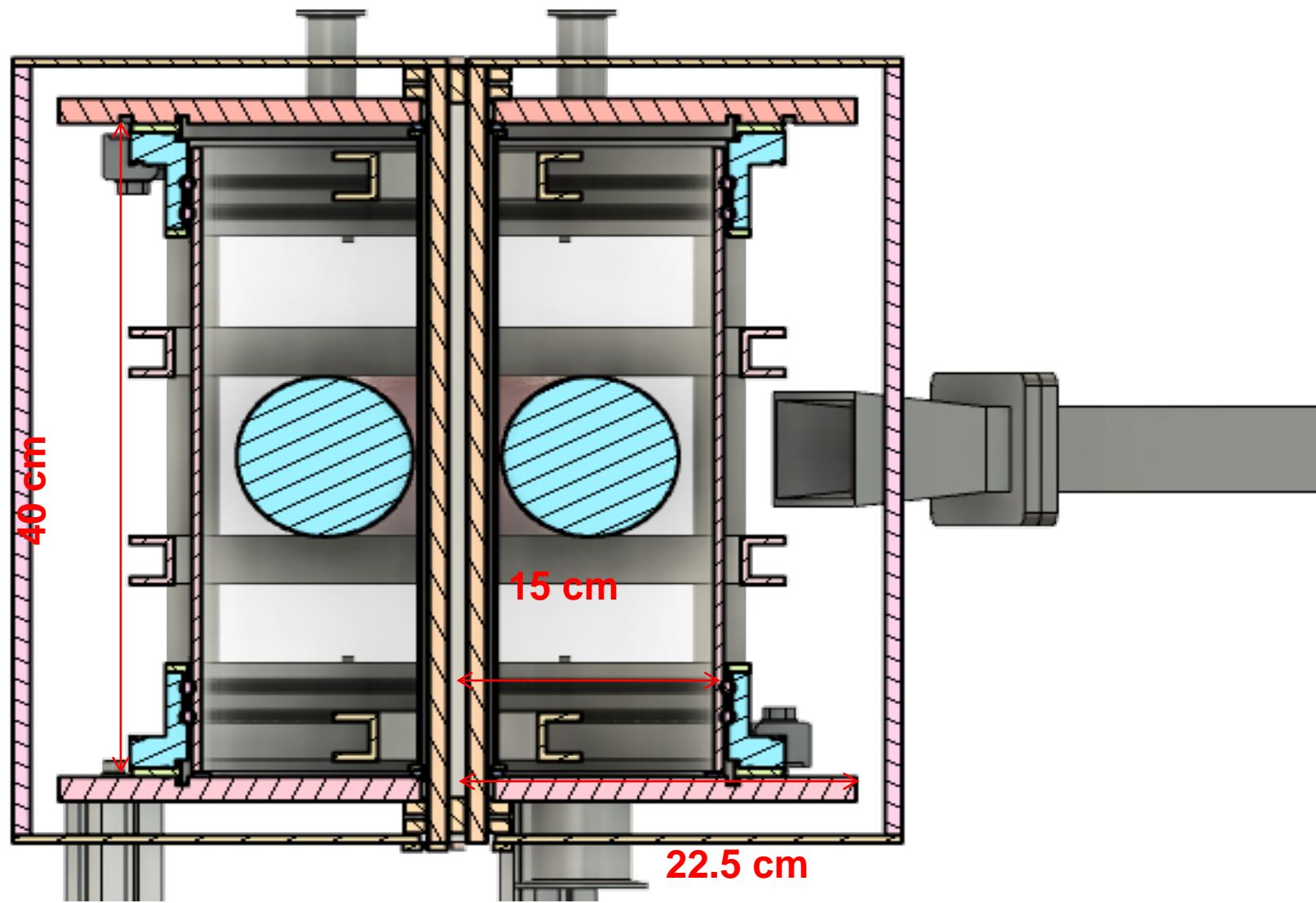
# 封合式視窗



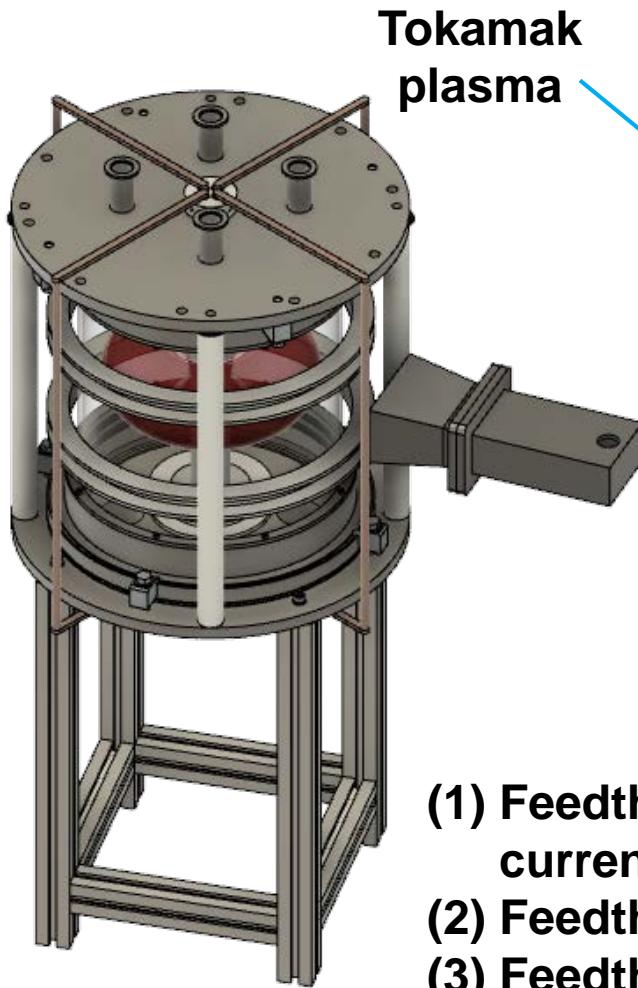
# 可拆式視窗



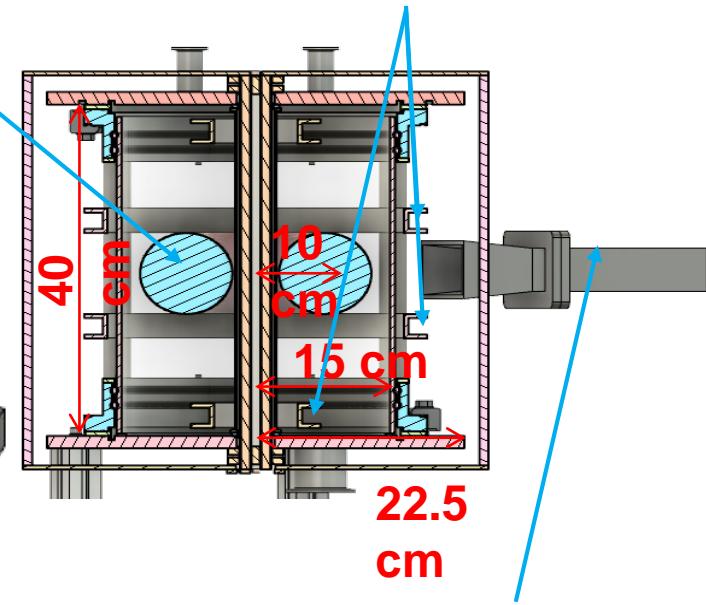
# Flange of the vacuum chamber for the spherical tokamak



# We need to work with a vacuum system



Vertical-field coils



1 kW, 2.45 GHz  
Magnetron

- (1) Feedthrough for conducting current to drive the vertical coil.
- (2) Feedthrough for Rogowski coil.
- (3) Feedthrough for triple probe.

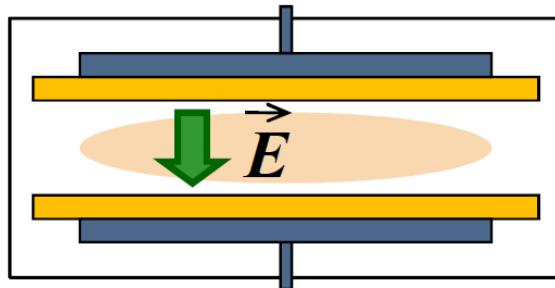
# Plasma will first be generated using capacitive coupling



- RF can interact with plasma inductively or capacitively.

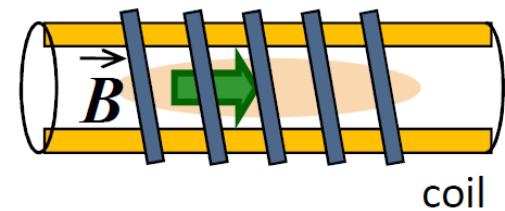
Capacitively coupled

planar



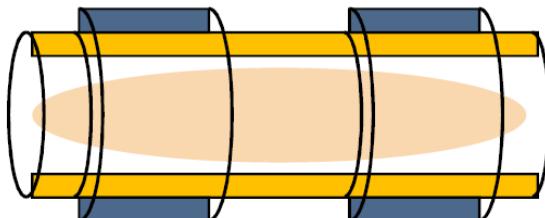
Inductively coupled

coaxial



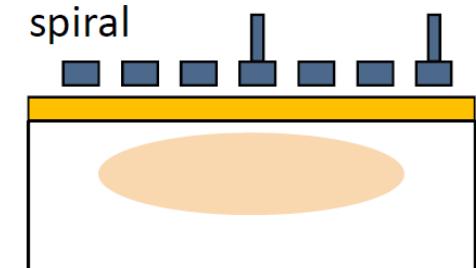
coil

coaxial

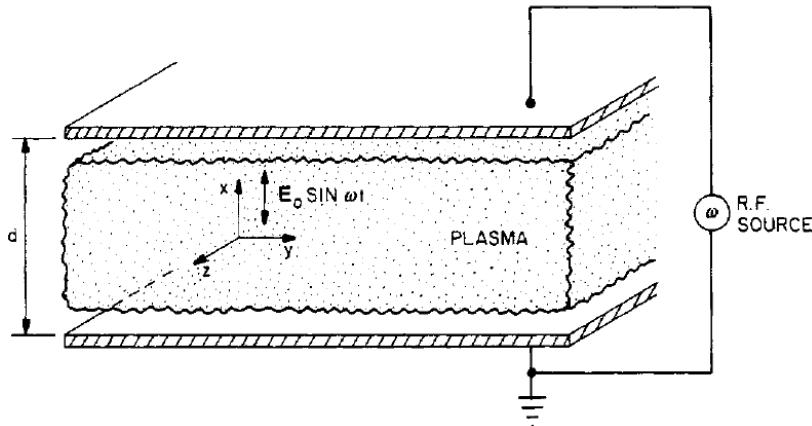


spiral

planar



# Capacitive RF coupling plasma without magnetic fields



$$\vec{F} = m \vec{a} = -\nu_c m \vec{v} - e \vec{E}$$

$$m \frac{dv_y}{dt} + m \nu_c v_y = 0$$

$$v_y(t) = v_{y0} \exp(-\nu_c t)$$

$$m \frac{d^2x}{dt^2} + m \nu_c \frac{dx}{dt} = e E_0 \sin(\omega t)$$

$$x = C_1 \sin(\omega t) + C_2 \cos(\omega t)$$

$$C_1 = -\frac{e E_0}{m} \frac{1}{\omega^2 + \nu_c^2}$$

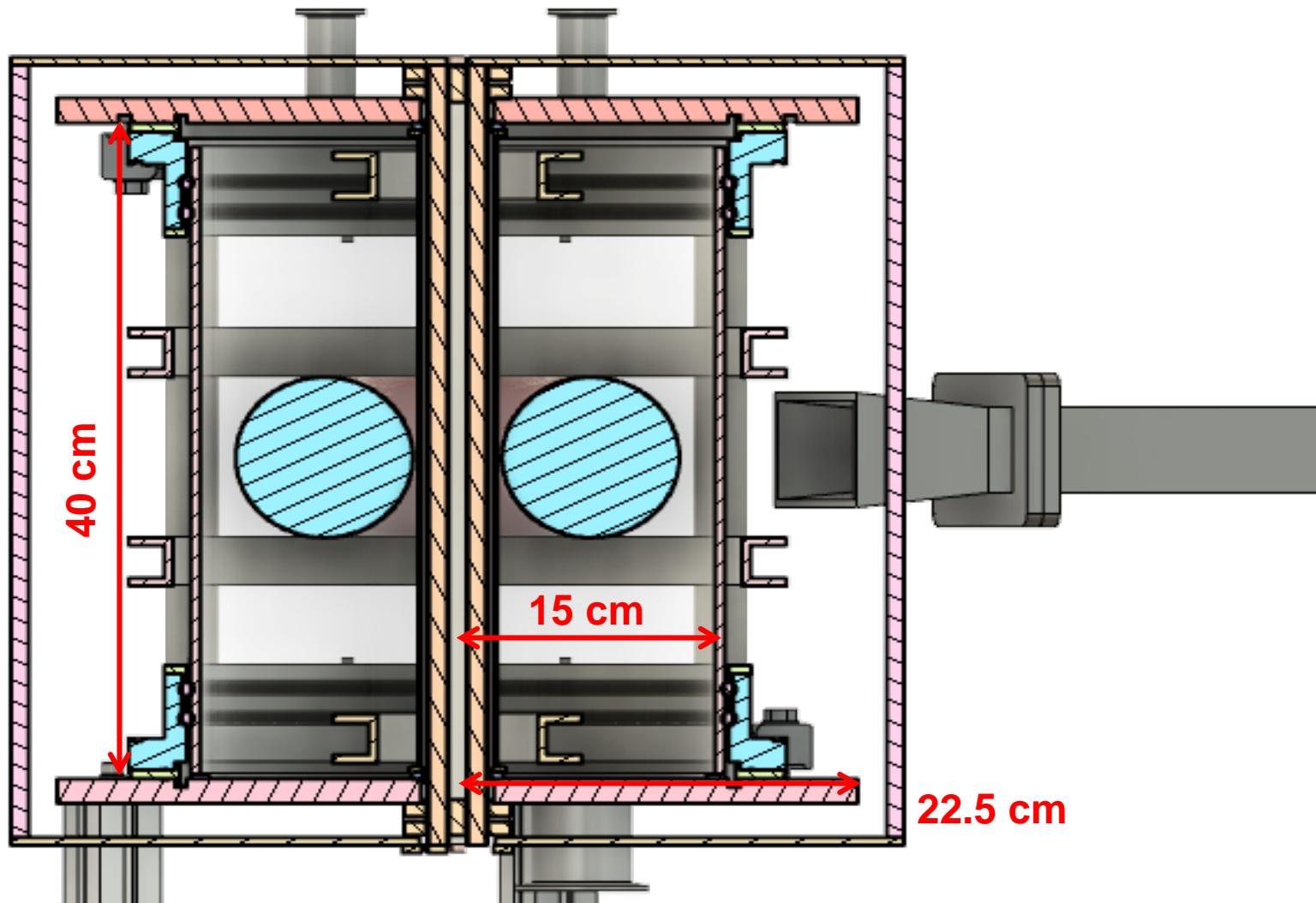
$$C_2 = -\frac{\nu_c e E_0}{\omega m} \frac{1}{\omega^2 + \nu_c^2}$$

$$v_x(t) = -\frac{e E_0 \omega}{m(\omega^2 + \nu_c^2)} \left[ \cos(\omega t) - \frac{\nu_c}{\omega} \sin(\omega t) \right]$$

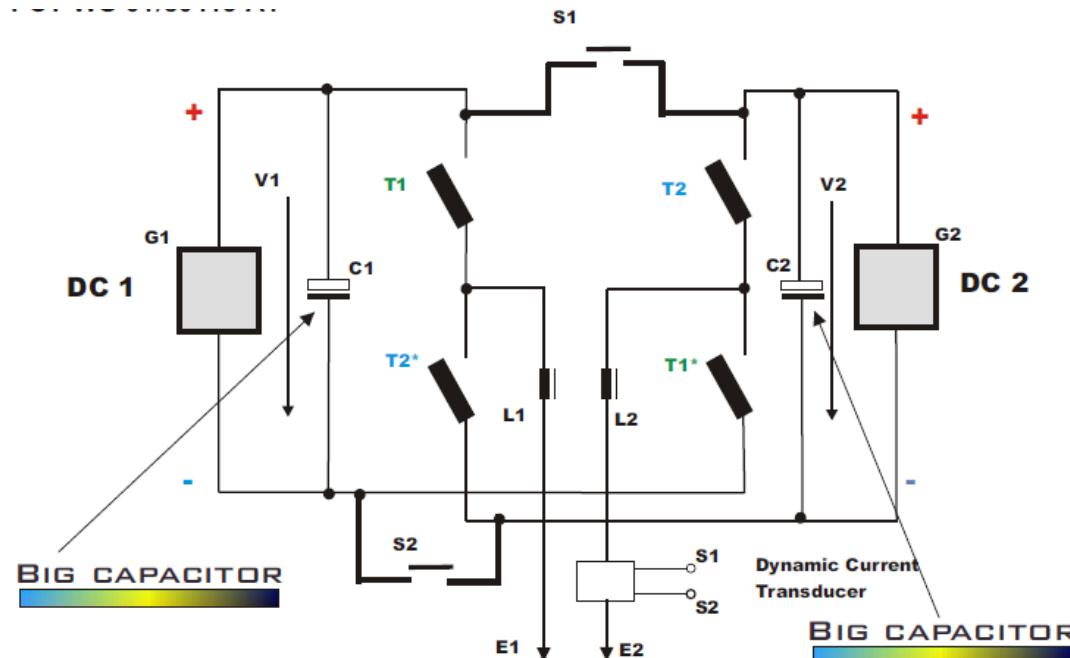
$$P = \frac{dW}{dt} = e E_0 \sin(\omega t) v_x$$

$$\bar{P}_{\text{tot}} = n_e \bar{P} = \frac{1}{4} \epsilon_0 E_0^2 \frac{2 n_e e^2}{m \epsilon_0} \frac{\nu_c}{\omega^2 + \nu_c^2}$$

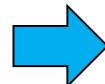
# High voltage electrode inserted from the feedthrough at the bottom of the chamber will be used for CCP



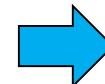
# A pulse generator will be used to convert a DC voltage to an AC voltage



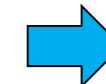
DC power supply



Pulse generator



Transformer



HV- RF output

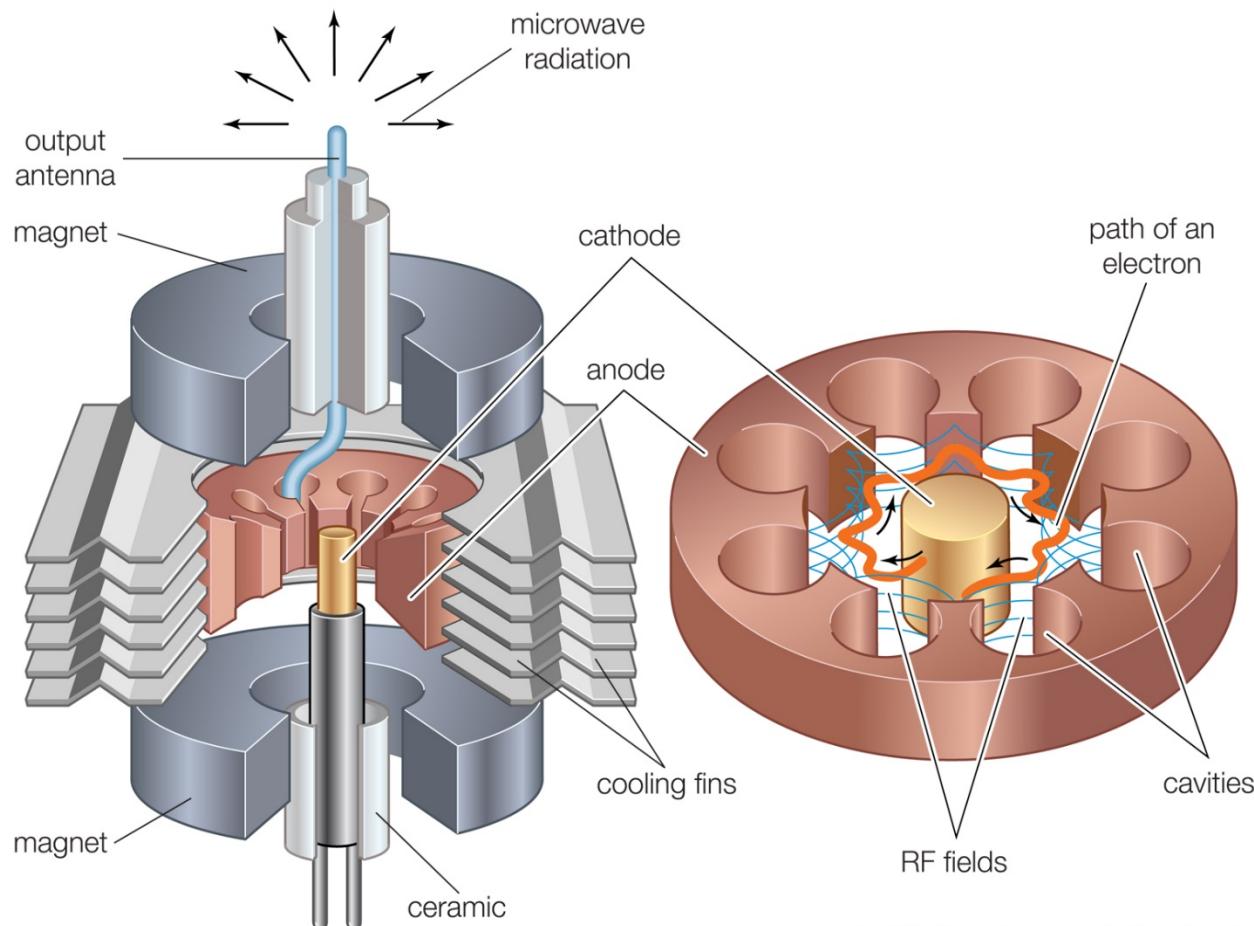
# Microwave heating

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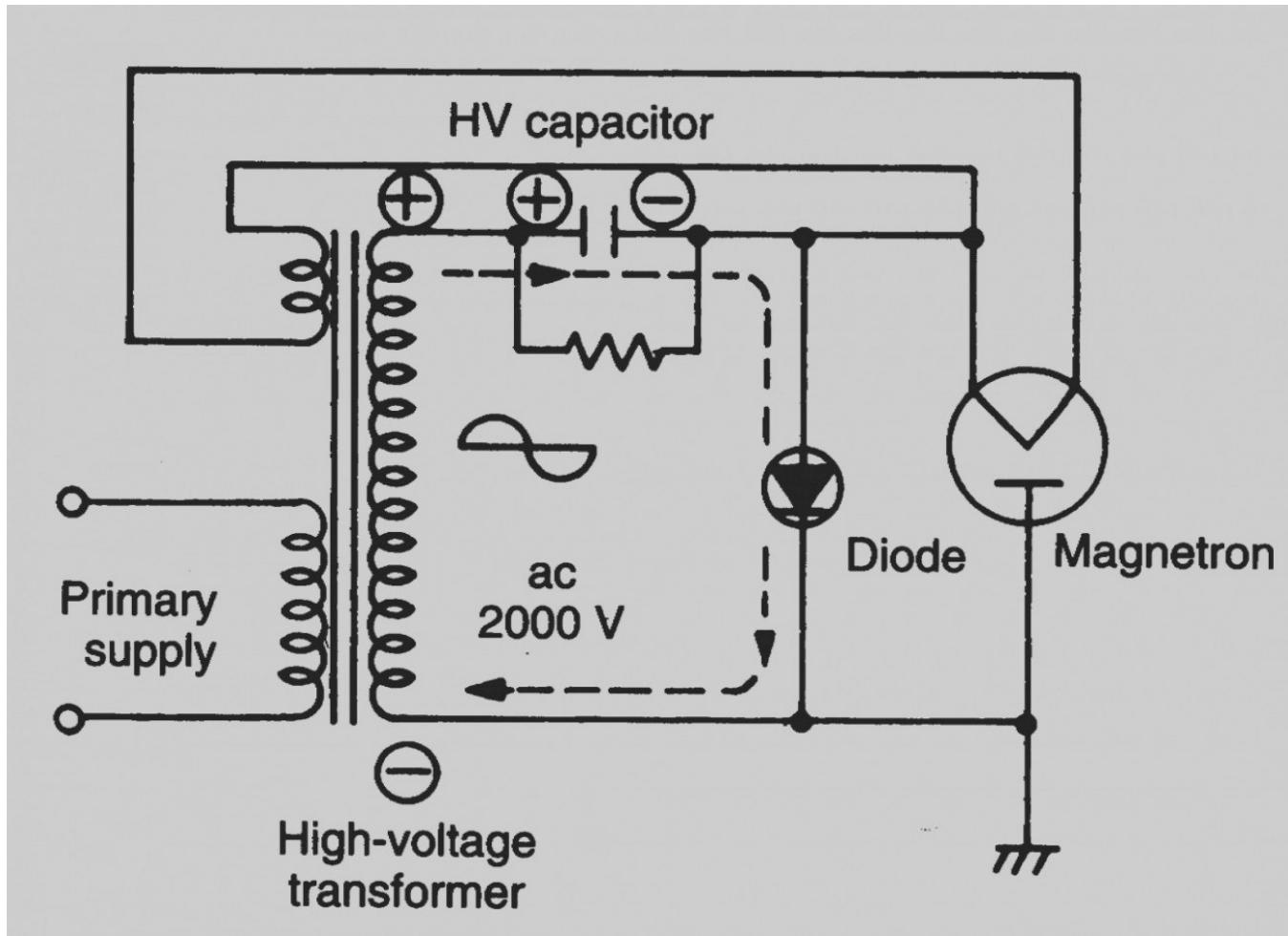
- **Microwave generation:**
  - Magnetron
  - Driver of magnetron
  - Microwave coupling
  - Microwave isolation
  - ECR heating – 1w, 2w, 3w
  - Required toroidal magnetic field -> coil design, current requirement
  - Required poloidal magnetic field, using safety factor
  - Required vertical field estimated from the density

# Internal of a magnetron

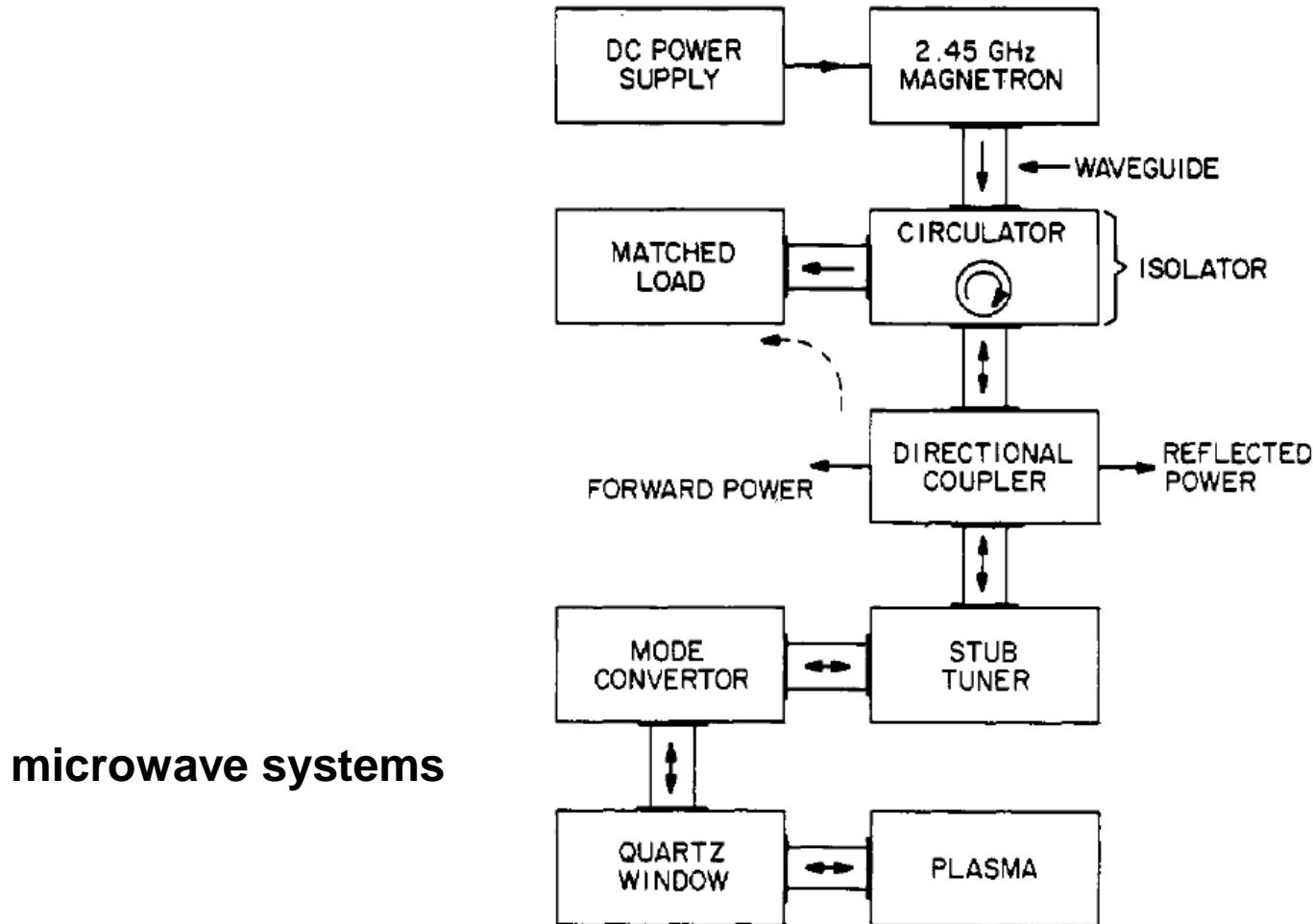


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# Magnetron schematic diagram



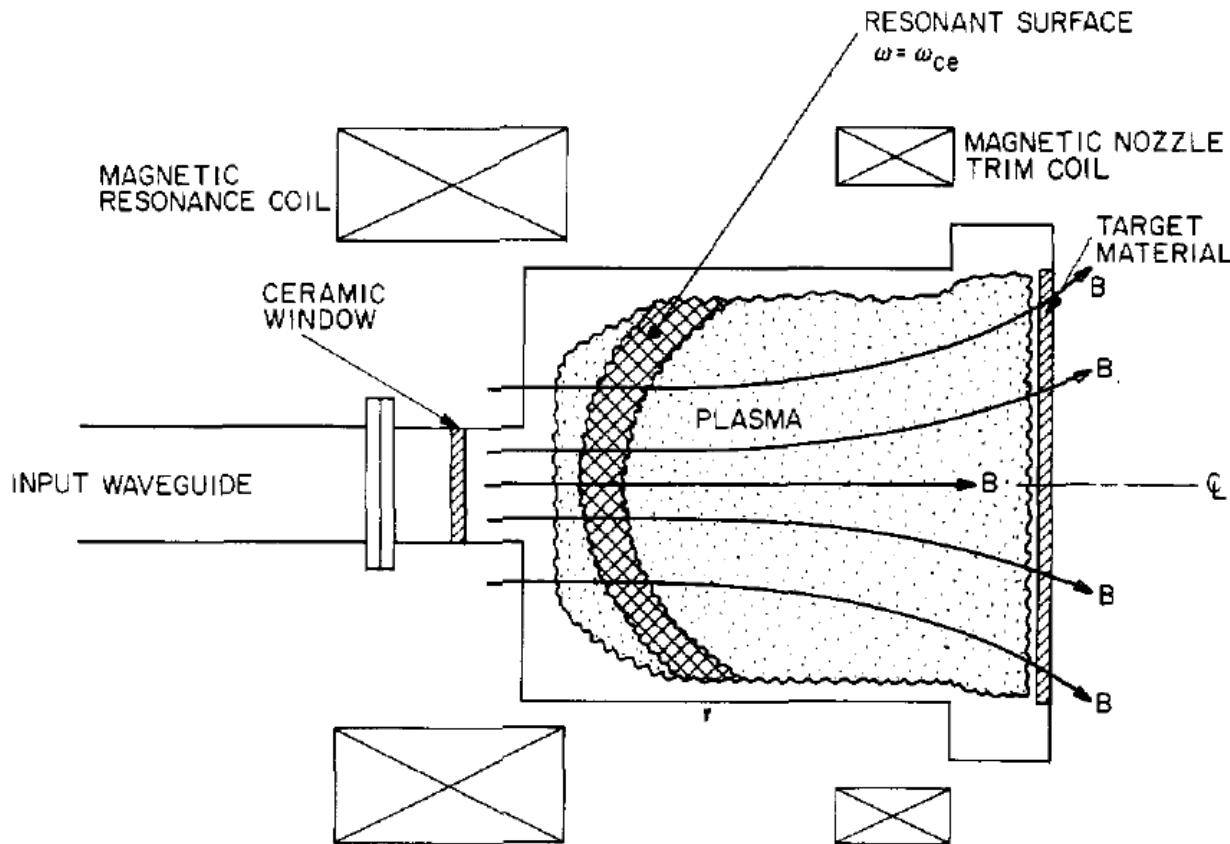
# Electron cyclotron resonance (ECR) microwave systems



# Strong absorption occurs when the frequency matches the electron cyclotron frequency



- Electron cyclotron resonance heating (ECH, ECR heating)



# Electron cyclotron frequency depends on magnetic field only



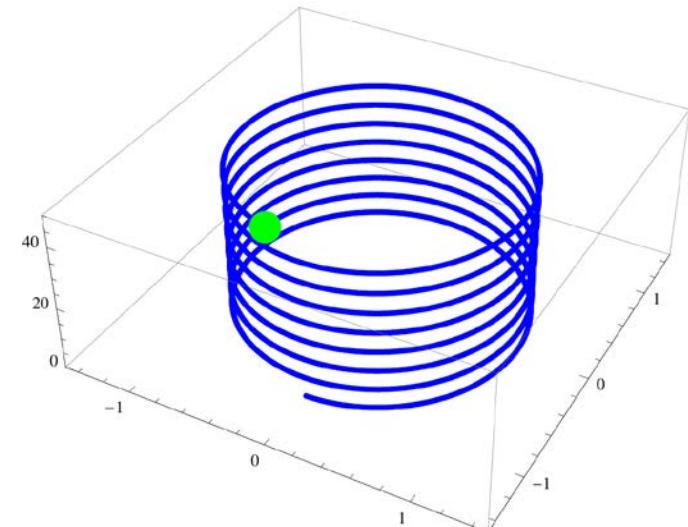
$$m_e \frac{d\vec{v}}{dt} = -\frac{e}{c} \vec{v} \times \vec{B}$$

- Assuming  $\vec{B} = B\hat{z}$  and the electron oscillates in x-y plane

$$\begin{aligned} m_e \dot{v}_x &= -\frac{e}{c} B v_y & m_e \dot{v}_z &= 0 \\ m_e \dot{v}_y &= \frac{e}{c} B v_x \end{aligned}$$

$$\ddot{v}_x = -\frac{eB}{m_e c} \dot{v}_y = -\left(\frac{eB}{m_e c}\right)^2 v_x$$

$$\ddot{v}_y = -\frac{eB}{m_e c} \dot{v}_x = -\left(\frac{eB}{m_e c}\right)^2 v_y$$



- Therefore

$$\omega_{ce} = \frac{eB}{m_e c}$$

# Electrons keep getting accelerated when a electric field rotates in electron's gyrofrequency



$$m_e \frac{d\vec{v}}{dt} = -\frac{e}{c} \vec{v} \times \vec{B} - e \vec{E}$$
$$\vec{B} = B_0 \hat{z} \quad \vec{E} = E_0 [\hat{x} \cos(\omega t) + \hat{y} \sin(\omega t)]$$

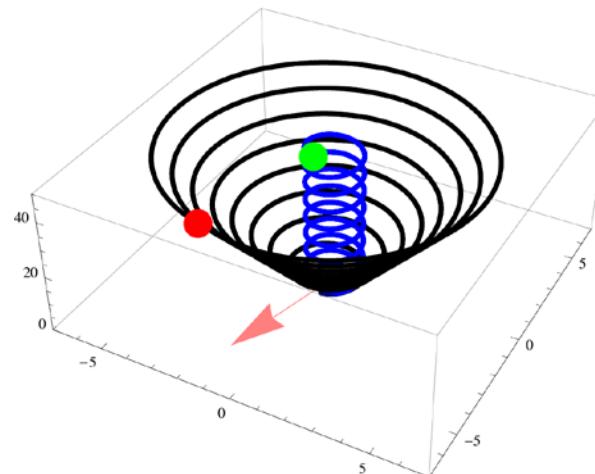
$$m_e \dot{v}_x = -\frac{e}{c} B v_y + E_0 \cos(\omega t)$$

$$m_e \dot{v}_y = -\frac{e}{c} B v_x + E_0 \sin(\omega t)$$

$$m_e \dot{v}_z = 0$$

$$\omega_{ce} = \frac{eB}{m_e c}$$

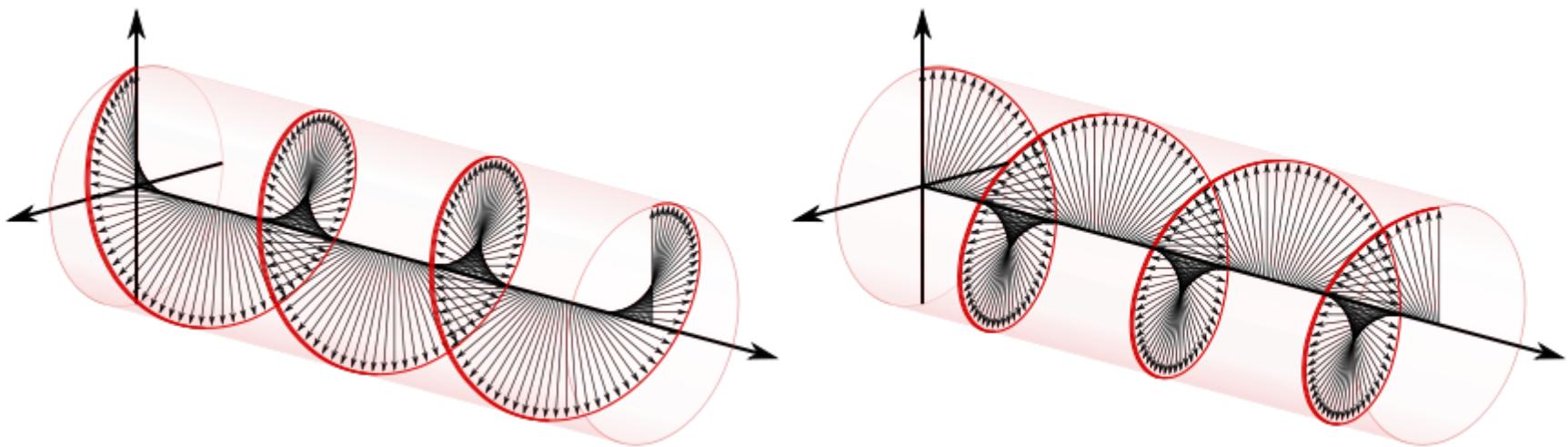
$$\ddot{v}_x = -\omega_{ce}^2 v_x - \frac{E_0}{m_e} (\omega_{ce} + \omega) \cos(\omega t)$$
$$\ddot{v}_y = -\omega_{ce}^2 v_y + \frac{E_0}{m_e} (\omega_{ce} + \omega) \sin(\omega t)$$



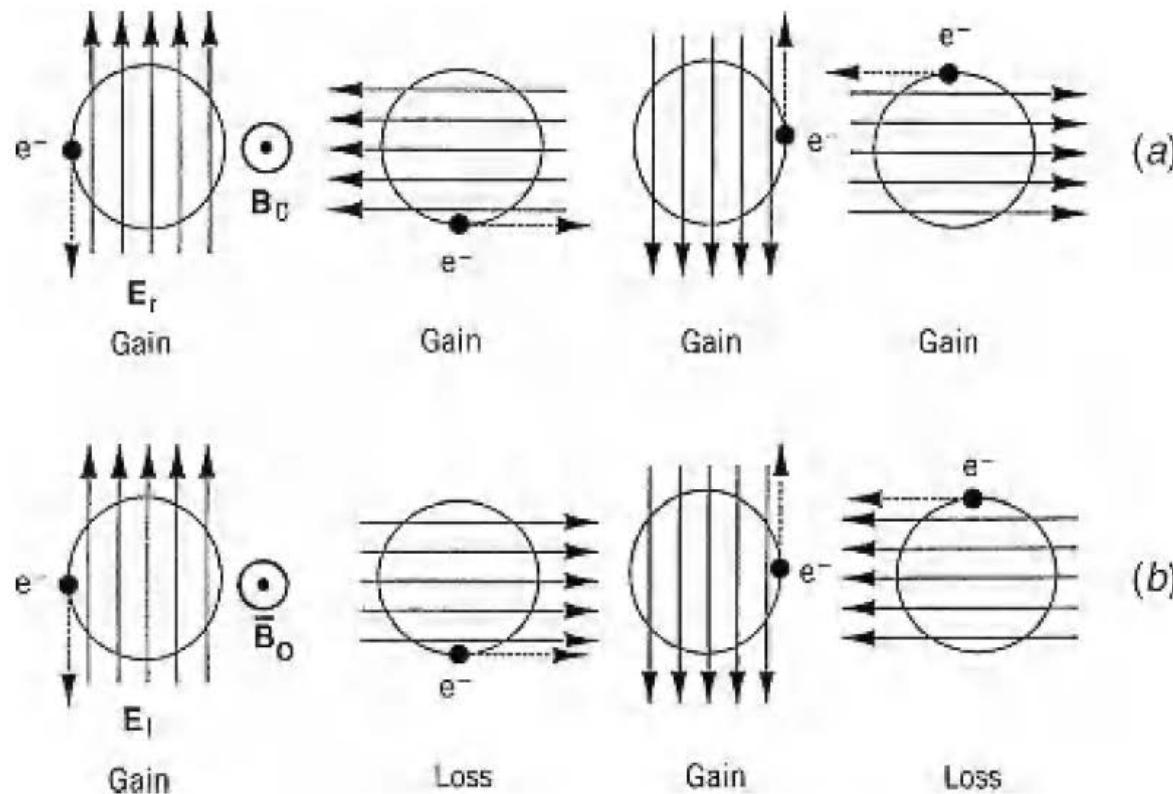
# Electric field in a circular polarized electromagnetic wave keeps rotating as the wave propagates



- Right-handed polarization
- Left-handed polarization



# Only right-handed polarization can resonance with electron's gyromotion



**FIGURE 13.5.** Basic principle of ECR heating: (a) continuous energy gain for right-hand polarization; (b) oscillating energy for left-hand polarization (after Lieberman and Gottscho, 1994).

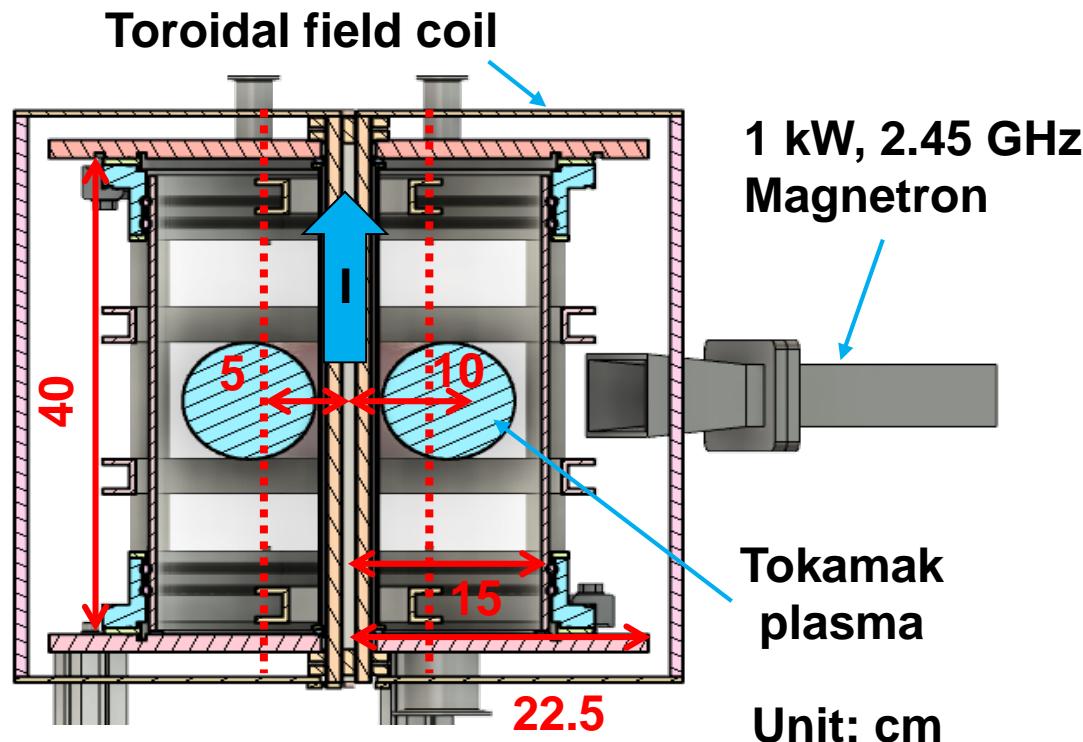
# A magnetic field of 0.0876 T is needed for ECH



$$\omega_{ce} = \frac{eB}{m_e c} \equiv \omega = 2\pi \times 2.45 \text{ GHz}$$

$$B = \frac{2\pi \times 2.45 \times 10^9 m_e c}{e} = 876 \text{ G} = 0.0876 \text{ T}$$

$$B = \frac{\mu_0 I}{2\pi r}$$



$$\begin{aligned} I &= \frac{2\pi r B}{\mu_0} = \frac{2\pi r B}{4\pi \times 10^{-7}} \\ &= 5B_{(T)}r_{(m)}(\text{MA}) \\ &= 22 \text{ kA} @ 5 \text{ cm} \end{aligned}$$

- A pulsed-power system will be used to generate the current.

# The 0.1-T magnetic field is sufficient to confine 10-eV Ar ion



$$v = r\omega \quad \frac{1}{2}mv^2 = kT \quad \omega = \frac{eB}{m} \quad r = \frac{\sqrt{2mkT}}{eB}$$

- Larmor radius in mm @ B=0.1 T:

T (eV)	H (1g/mole)	D (2g/mole)	T (3g/mole)	He (4g/mole)	Ar (40g/mole)
1	1.4	2.0	2.5	2.9	9.1
10	4.6	6.5	7.9	9.1	28.9
100	14.4	20.4	25.0	28.9	91.3
1000	45.6	64.5	79.1	91.3	288.7

- The Larmor radius of 1-keV electron @ B=0.1 T is 1.1 mm.

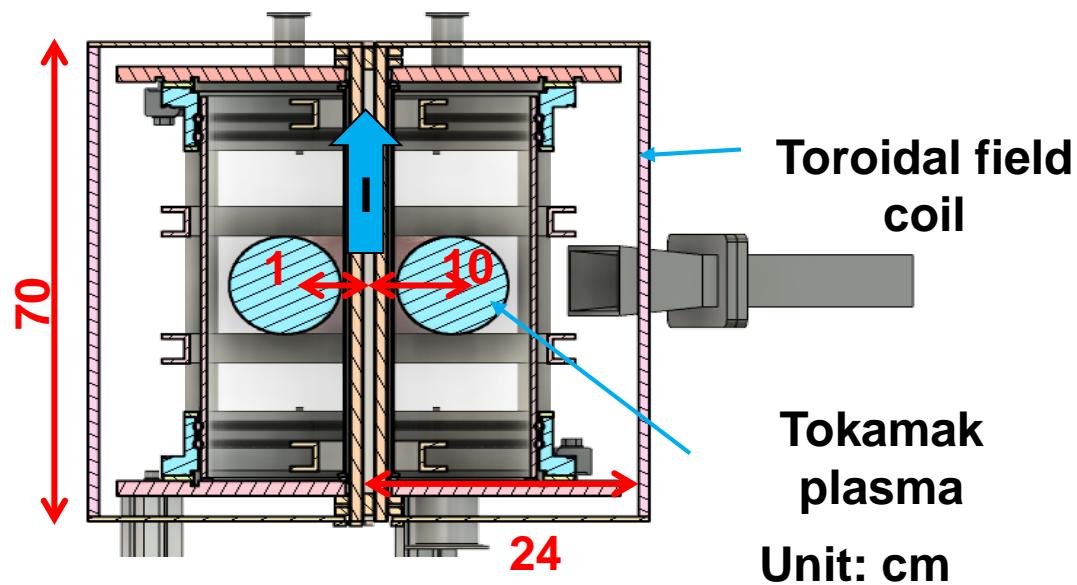
• Ar will be used.

# The magnetic field energy of the toroidal field is ~100 J



$$B = \frac{\mu_0 I}{2\pi r}$$

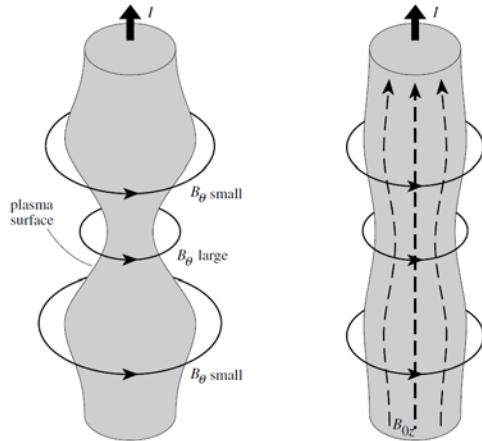
$$\begin{aligned} E &= \int \frac{B^2}{2\mu_0} dv = \int_{r_{\min}}^{r_{\max}} \frac{B^2}{2\mu_0} 2\pi r L dr = \frac{2\pi L}{2\mu_0} \frac{\mu_0^2 I^2}{(2\pi)^2} \int_{r_{\min}}^{r_{\max}} \frac{1}{r^2} r dr = \frac{\mu_0 L I^2}{4\pi} \int_{r_{\min}}^{r_{\max}} \frac{1}{r} dr \\ &= \frac{\mu_0 L I^2}{4\pi} \ln\left(\frac{r_{\max}}{r_{\min}}\right) = 10^{-7} \times 0.7 \times (20 \times 10^3)^2 \ln\left(\frac{24}{1}\right) \\ &= 89 \text{ J} \end{aligned}$$



# Instabilities occur in a cylindrical plasma column



- Sausage instability:



- Safety factor  $q$ :

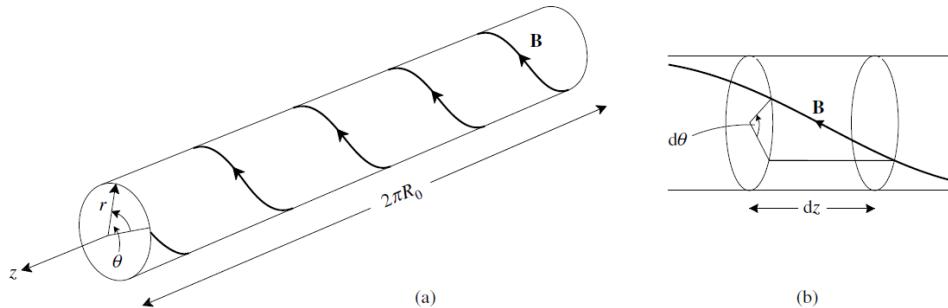
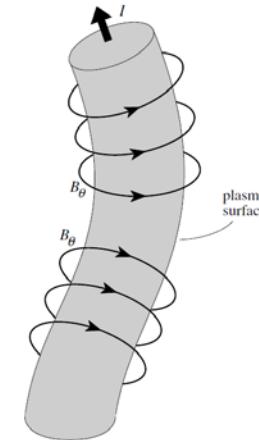


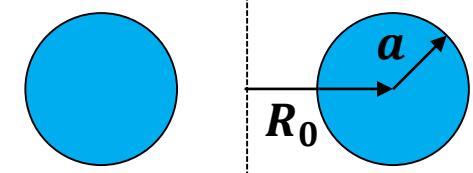
Fig. 4.6. Screw pinch geometry.

- Kink instability:



$$q(r) = \frac{rB_r(r)}{R_0 B_\theta(r)} \approx \frac{rB_t}{R_0 B_p}$$

$(R_0 \gg a)$



# A plasma current of $\sim 2$ kA is needed

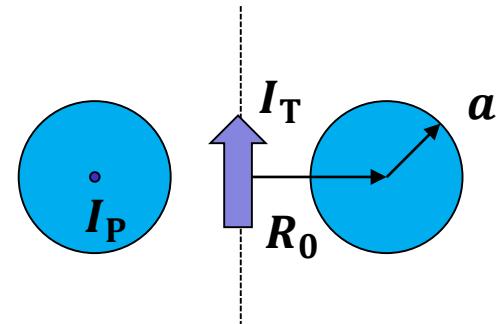


$$q(r) \approx \frac{rB_t}{R_0B_p} \approx \frac{aB_t}{R_0B_p}$$

$$B_T = \frac{\mu_0 I_T}{2\pi R_0} \quad B_P = \frac{\mu_0 I_p}{2\pi a}$$

$$I_p \sim \frac{1}{q} \left( \frac{a}{R_0} \right)^2 \quad I_T = \frac{1}{3} \left( \frac{5}{10} \right)^2 \quad 20\text{kA} \sim 2\text{kA}$$

$$B_p \sim 40 \text{ G}$$

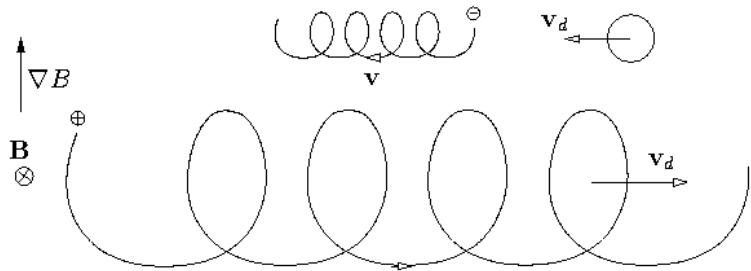


$$R_0 \sim 10 \text{ cm}$$
$$a \sim 5 \text{ cm}$$

# Plasma current will be generated by the Grad-B drift and the Curvature drift current

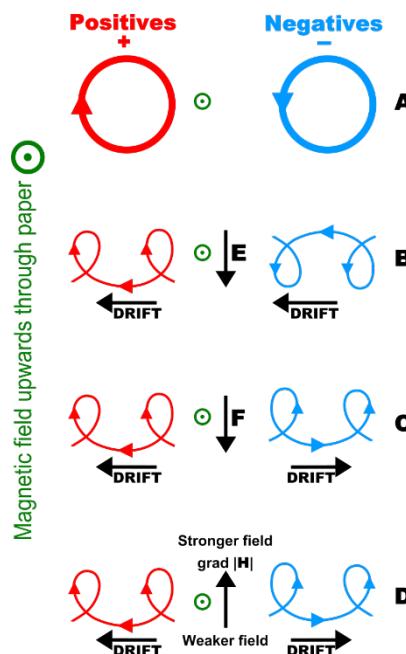
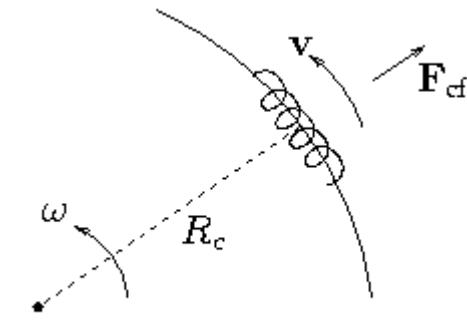


- Grad-B drift



$$V_{\nabla B} = \pm \frac{1}{2} v_{\perp} r_L \frac{\vec{B} \times \nabla B}{B^2}$$

- Curvature drift



$$V_R = \frac{mv_{||}^2}{q} \frac{\vec{R}_c \times \vec{B}}{R_c^2 B^2}$$

$$V_R + V_{\nabla B} = \frac{m}{q} \left( v_{||}^2 + \frac{1}{2} v_{\perp}^2 \right) \frac{\vec{R}_c \times \vec{B}}{R_c^2 B^2}$$

$$\approx \frac{1}{q} (2T_{||} + T_{\perp}) \frac{\vec{R}_c \times \vec{B}}{R_c^2 B^2}$$

# A vertical field $B_v$ of 12 G with a curvature of 5 cm is needed to generate the required plasma current

- For  $P = 10^{-1}$  Torr = 13 Pa = 13 N/m<sup>2</sup>

$$n = \frac{P}{T} = \frac{13}{4.1 \times 10^{-21}} = 3.1 \times 10^{21} \text{ m}^{-3}$$

Assuming the ionization fraction is 1%:

$$n_e = n_i = 3.1 \times 10^{19} \text{ m}^{-3}$$

$$j = qn_e v \quad I \sim \pi a^2 j = \pi a^2 qn_e v$$

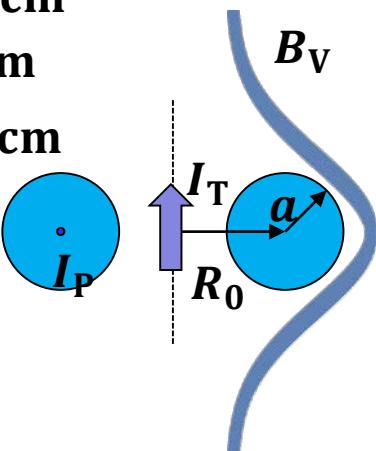
$$v = \frac{I}{\pi a^2 qn_e} = \frac{2 \times 10^3}{\pi 0.05^2 \times 1.6 \times 10^{-19} \times 3.1 \times 10^{19}} \sim 5 \times 10^4 \text{ cm/s}$$

$$v_{\text{drift}} = V_R + V_{VB} \approx \frac{1}{q} (2T_{||} + T_{\perp}) \frac{\vec{R}_c \times \vec{B}_V}{R_c^2 B_V^2} \sim \frac{3T}{q} \frac{1}{R_c B_V}$$

$$B_V \sim \frac{1}{V_{\text{drift}}} \frac{3T}{q} \frac{1}{R_c} \sim \frac{1}{5 \times 10^4} 3 \times 1 \times \frac{1}{0.05} \sim 0.0012 \text{ T} = 12 \text{ G} \quad (B_{\text{earth}} \sim 0.5 \text{ G})$$

- For  $T_e = 1 \text{ eV}$ ,  $B_v = 12 \text{ G}$

$$r_c = \frac{\sqrt{2mT}}{eB_V} = \frac{\sqrt{2 \times 9.11 \times 10^{-31} \times 1.6 \times 10^{-19}}}{1.6 \times 10^{-19} \times 0.0012} = 2.8 \text{ mm}$$



# The prospective system design



- (1) Vertical field coil (VF coil):  $B_v = 12 \text{ G}$  w/ curvature of 5 cm.
- (2) Pulse forming network for driving VF coil: ? kA.
- (3) Rogowski coil for measuring plasma current:  $I_p = 2 \text{ kA}$ .
- (4) Triple probe for measuring Plasma characteristics:  $T_e \sim 1 \text{ eV}$ ,  $n_e \sim 10^{19} \text{ m}^{-3}$ .

